TECHNICAL MANUAL

PHASED MAINTENANCE INSPECTION CHECKLIST FOR

ARMY AH-64D HELICOPTER

DISTRIBUTION STATEMENT A: Approved for public release; distribution is unlimited.

Note: This Technical Manual supports AH-64D aircraft serial numbers 96-5001 through 01-5284

HEADQUARTERS, DEPARTMENT OF THE ARMY 24 MAY 2002

NO. 1

HEADQUARTERS DEPARTMENT OF THE ARMY WASHINGTON, D.C., 16 October 2003

PHASED MAINTENANCE INSPECTION CHECKLIST

FOR

AH-64D HELICOPTER

DISTRIBUTION STATEMENT A: Approved for public release; distribution is unlimited

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nemove pages	insert pages
A and B	A and B
1-1 through 1-4	1-1 through 1-4
2-1 through 2-6	2-1 through 2-6
2-9 and 2-10	2-9 and 2-10
2-15 through 2-23/(2-24 blank)	2-15 through 2-23/(2-24 blank)
2-25 through 2-32	2-25 through 2-32
2-35 through 2-37/(2-38 blank)	2-35 through 2-37/(2-38 blank)
2-43 through 2-50	2-43 through 2-50
2-53 through 2-65/(2-66 blank)	2-53 through 2-65/(2-66 blank)
2-67 through 2-71/(2-72 blank)	2-67 through 2-71/(2-72 blank)
2-73 and 2-74	2-73 and 2-74
2-79 through 2-96	2-79 through 2-96
2-103 through 2-105/(2-106 blank)	2-103 through 2-105/(2-106 blank)
2-107 through 2-115/(2-116 blank)	2-107 through 2-115/(2-116 blank)
2-117 through 2-123/(2-124 blank)	2-117 through 2-123/(2-124 blank)

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Page	*Change	Page	*Change
No.	No.	No.	No.
Cover	0	2-36–2-37	1
A–B		2-38 blank	
Title		2-39 through 2-41	
1-1 through 1-4		2-42 blank	
1-5 through 1-2		2-43 through 2-46	
2-1		2-47	
2-2 and 2-3		2-48 through 2-50	
2-4		2-51	
2-5 and 2-6		2-51	
2-7 and 2-8		2-52 blank	
2-9		2-55	
2-10 through 2-		2-56	
2-14 blank		2-57	
2-14 blank 2-15 and 2-16		· · · · · · · · · · · · · · · · ·	
		2-58 through 2-65	
2-17		2-66 blank	
2-18 through 2-		2-67	
2-21	-		
2-22 and 2-23		2-70	
2-24 blank		2-71	
2-25		2-72 blank	
2-26 and 2-27		2-73	
2-28		2-74 through 2-78	
2-29 through 2-		2-79 through 81	
2-33 through 2-	-350	2-82 and 2-83	0

^{*}Zero in this column indicates an original page.

Page	*Change	Page	*Change
No.	No.	No.	No.
2-84 and 2-89	1	2-109	0
2-90	0	2-110 and 2-111	1
2-91 through 2-93	31	2-112	0
2-94	0	2-113 through 2-115.	1
2-95 and 2-96	1	2-116 blank	0
2-97 through 2-10	030	2-117 through 2-120	1
2-104 and 2-105	1	2-121	0
2-106 blank	0	2-122 and 2-123	1
2-107 and 2-108	1	2-124 blank	0

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HEADQUARTERS DEPARTMENT OF THE ARMY WASHINGTON, D.C., 24 MAY 2002

PHASED MAINTENANCE INSPECTION CHECKLIST FOR

ARMY AH-64D HELICOPTER

DISTRIBUTION STATEMENT A: Approved for public release; distribution is unlimited.

WARNING

CERTAIN INSPECTIONS ARE MANDATORY SAFETY-OF-FLIGHT REQUIREMENTS, AND THE INSPECTION INTERVALS CANNOT BE EXCEEDED. IN THE EVENT THESE INSPECTIONS CANNOT BE ACCOMPLISHED AT THE SPECIFIED INTERVAL, THE AIRCRAFT CONDITION STATUS SYMBOL WILL BE IMMEDIATELY CHANGED TO A RED X. MANDATORY SAFETY-OF-FLIGHT INSPECTION ITEMS ARE PRINTED IN BOLD FACE TYPE.

NOTE

INSPECTION ITEMS CONTAINED IN THIS MANUAL ARE CONSIDERED THE MINIMUM REQUIREMENTS FOR PERFORMING PHASED MAINTENANCE AND MUST BE PERFORMED. THE CUMULATIVE EFFECTS OF INSPECTION DEFERRALS ARE UNKNOWN AND COULD RESULT IN CATASTROPHIC FAILURE OR INCREASED MAINTENANCE AT A LATER DATE. THEREFORE, THE USE OF SPECIAL LETTERING TO EMPHASIZE MANDATORY SAFETY-OF-FLIGHT INSPECTION ITEMS IS NOT TO BE CONSTRUED AS AUTHORITY FOR DEFERRAL OF OTHER INSPECTIONS.

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SECTION I. GENERAL INFORMATION

PHASED SCHEDULE. This phased maintenance inspection checklist contains requirements for inspection of the AH-64D helicopter on a phased schedule having a 1000-hour (flight hours) cycle with 250-hour phases. Each requirement included herein is designated for accomplishment at least once, but not more than four times during the 1000-hour cycle.

EXCEEDING THE PHASED SCHEDULE. The phased maintenance inspection intervals designated are the maximum and shall not be exceeded except in actual operational emergencies as explained herein. It is the Commander's responsibility to determine (on an individual aircraft basis) when inspection intervals may be exceeded. For this purpose, operational emergencies are conditions of combat, or conditions of disaster which necessitate flight to evacuate aircraft or personnel. Those inspections annotated by a "C" in the Inspect Phase Nos. column, along with the DA Form 2408-18 (Equipment Inspection List), are considered the MINIMUM mandatory combat maintenance inspection requirements for helicopters scheduled for imminent deployment to or stationed in a combat environment. Under no circumstances will two combat maintenance inspections be performed sequentially. When aircraft are operated beyond the normal inspection due time because of such emergency situations, a circled red X status symbol and an appropriate statement (to include authority) must be entered in block 16 and 17 of DA Form 2408-13-1 (Aircraft Inspection and Maintenance Record) until such time as the inspection is complete. When inspections are delayed to meet emergency requirements, Commanders will assure that the aircraft status symbol reverts to a red X and that delayed inspections are accomplished immediately upon termination of the emergency. When unusual local conditions (utilization, type of mission, personnel, periods of inactivity, environmental conditions, etc.) dictate, it is the prerogative and responsibility of the Maintenance Officer to increase the scope and/or frequency of maintenance or inspection as necessary to ensure safe operation (TM 1-1500-328-23).

MAINTENANCE ACTIVITIES. The inspections prescribed by this checklist will be accomplished at specified phases by Aviation Unit Maintenance (AVUM) activities with assistance of Aviation Intermediate Maintenance (AVIM) and Depot Maintenance activities when required. The inspection of the part/component is visual unless stated otherwise.

LIMITATIONS. The checklist does not contain instructions for repair, adjustment, or other means of rectifying conditions. Neither does it contain special tolerances, limits, or instructions for special troubleshooting to find causes for malfunctions. Such data will be obtained from the latest issue of the aircraft TM 1-1520-Longbow/ Apache IETM.

CHANGEOVER TO THE PHASED MAINTENANCE SYSTEM. Changeover shall be accomplished in accordance with instructions provided in TB 55-1500-337-24 entitled, "Phased Maintenance System for Army Aircraft." The requirements of this TB must be accomplished prior to implementation of Phase 1 inspection requirements specified in this checklist.

PRE-INSPECTION MAINTENANCE TEST FLIGHT (MTF). A pre-inspection MTF to duplicate non-hazardous equipment problems, determine unsatisfactory conditions, determine equipment operation problems, etc., is recommended prior to start of aircraft disassembly for phased maintenance inspection. The decision to perform the pre-inspection MTF, however, shall be the responsibility of the unit Maintenance Officer.

SPECIAL INSPECTIONS, CALENDAR INSPECTIONS, AND LUBRICATION REQUIREMENTS. Special inspections, calendar inspections, and lubrication requirements contained in TM 1-1520-Longbow/Apache and those listed on the aircraft's DA Form 2408-18 shall be reviewed and accomplished in accordance with the "inspection due" requirements specified in those documents.

TIME BETWEEN OVERHAUL (TBO) AND RETIREMENT LIFE ITEMS CHECK. Prior to start of the applicable phased maintenance inspections and lubrication, a check will be made of components and their remaining operating hours prior to removal. The latest issue of the aircraft's TM 1-1520-Longbow/Apache and DA Form 2408-16 shall be referred to for a complete listing of components and their TBO and retirement life.

USING THE PHASED INSPECTION CHECKLIST.

- a. A new checklist shall be used each time phased maintenance is due on the aircraft. This checklist is arranged such that it can be separated by area and distributed to the maintenance crew.
 - (1) Space is provided on each checklist form for entering the following data:
 - (a) The number of the maintenance inspection being performed.
 - (b) Aircraft serial number.
 - (c) Date of inspection.
 - (d) Total hours. (Block provided for local use.)
 - (2) For each inspection item a column is provided for entering the following data:
 - (a) Status of the aircraft as the result of the inspection requirement.
 - (b) Aircraft fault and/or remarks indicated by the inspection requirement.
 - (c) Action taken to correct the fault.
 - (d) Initials of person performing the corrective action.
- b. This checklist is formatted to eliminate the requirements to use DA Form 2408-13-3 as temporary records during phased inspections. Figures 1 thru 3 show examples of methods used to make entries on the phased maintenance checklist forms and the use of supplemental sheets for continuation purposes. This checklist pertains to all AH-64D helicopters and may, therefore, contain inspection requirements applicable to specific equipment not installed on individual aircraft. When this situation is encountered, those requirements that are not applicable need not be performed.
- c. A supplemental Checklist Sheet form (DA Form 4676-R) (figure 3, Sheet 2) provided at the end of Section I of this checklist is to be used for local reproduction. Copies of this form will be used to write up faults, remarks, and corrective actions when additional space is required. These supplemental sheets will be used instead of DA Form 2408-13-3 in the accomplishment of the phased maintenance inspections.
 - d. Faults and remarks on the DA Form 2408-13-1 and DA Form 2408-14 may be transcribed to this checklist at the discretion of the unit Maintenance Officer.
- PHASE NUMBERS. In the column headed "Inspect Phase Nos." and adjacent to the sequence number of each inspection requirement, there will appear the word "ALL" or a series of numbers. The word "ALL" indicates that inspection requirement shall be accomplished at each phase (or at every 250-hour interval) of the 1000-hour cycle. The numbers represent the phase number at which that inspection requirement is to be accomplished. For example, if the numbers 2 and 4 are shown, that inspection requirement is to be accomplished at phases 2 and 4 only (or at 500-hour interval). If only one number is indicated, then that inspection requirement is accomplished at that phase (or at every 1000-hour interval). At the completion of phase 4, the cycle starts over again with Phase 1.

STATUS SYMBOLS. All faults and deficiencies discovered during the inspection will be recorded on DA Form 2408-13-1/2408-13-1-E. The status symbols used are the same as those defined in DA PAM 738-751. The status symbol shall be entered by the person(s) performing the inspection and is determined by the type of fault that is found. Do not enter a horizontal dash (-) on the checksheet merely to show a particular inspection requirement is due. If an inspection reveals no fault, a status symbol will not be entered. The person clearing the fault shall place his last name initial over the status symbol. A red X or a circled red X symbol will not be initialed over until after the corrective action has been approved and signed off by a Technical Inspector or designated supervisor.

FAULTS AND/OR REMARKS. Fault entries in the Faults and/or Remarks column shall be brief remarks which describe the conditions resulting from the inspection and which require corrective action. The initials of the person making the entry will be entered immediately after the entry. If no fault is found, this column will be left blank.

ACTION TAKEN.

- a. Entries in the Action Taken column shall be brief remarks which describe the action taken to correct the fault described in the adjacent Faults and/or Remarks column. When faults are assigned a red X status, the corrective action shall be inspected and signed off by the Technical Inspector or designated supervisor.
- b. If no fault was found, an appropriate remark shall be entered in this column to indicate that the inspection was accomplished, e. g., "Inspected and found OK." If an inspection item is not applicable to the particular inspection phase number in work or to specific equipment installed on an individual aircraft, a "N/A" entry is required. The initials of the person making the entry shall be entered in the Initial column.

INITIAL. The person correcting the indicated fault shall enter his initials in the initial column opposite the first line of the Action Taken entry.

FINAL RECORDS CHECK. After all corrective actions have been completed and following completion of the phased inspection, the Technical Inspector or designated supervisor shall verify that all applicable forms and records have been properly updated. All uncorrected faults shall be entered on DA Form 2408-13-1, prepared for that date or to the DA Form 2408-14. A Final Records Checklist (table I) is provided to ensure forms and records have been inspected for completeness and accuracy prior to release of the aircraft from the phased maintenance inspection. The inspector verifying the final records check shall enter his initials adjacent to the indicated form on the Final Records Checklist. The initials entered shall be registered on the Signature Sheet (table II) adjacent to that person's signature. Once the final records check is complete, provide electronic or hard copies of 2408-16 and 2408-16-1 to: TAMMS-A@redstone.army.mil or Commander, U.S. Army Aviation and Missile Command, ATTN: AMASM-MMC-MA-NC, Redstone Arsenal, AL, 35898-5230.

SIGNATURE SHEET. All personnel performing inspection and/or maintenance tasks shall place their signatures and initials on the signature sheet (table II). The purpose of the signature sheet is to provide a correlation between initials entered on the individual checklist sheets and the actual names of the personnel accomplishing these tasks.

MAINTENANCE OPERATIONAL CHECKS. After the completion of any required corrective actions to any of the components of a functional system of the aircraft, maintenance operational checks (MOC) shall be performed on that system to determine the effectiveness of the maintenance actions performed and to verify the proper operation of that system. These MOC shall be performed in accordance with TM 1-1500-328-23. Copies of supplemental sheets (DA Form 4676-R) may be used to record and sign off the Maintenance Operational Checks performed.

MAINTENANCE TEST FLIGHT. When all required inspections in Section II have been accomplished and initialed in accordance with the above procedure, a daily inspection in accordance with the TM specified in Section II will be performed on the aircraft to permit performance of a maintenance test flight (MTF). The MTF shall be performed in accordance with the requirements of TM 1-1520-Longbow/Apache and TM 1-1500-328-23 using the MTF form in the MTF technical manual. A suggested maintenance test flight checksheet (figure 4) and a rotor smoothing record (figure 5) are provided at the end of Section I.

CHECKLIST DISPOSITION. The completion of each phased maintenance inspection shall be recorded on DA Form 2408-13-1 and 2408-15 as prescribed by DA PAM 738-751. The signed checklist, together with all continuation sheets, shall be attached to DA Form 2408-13-1 and filed for the six months period as required by DA PAM 738-751. At the end of the six months period, records will be destroyed per disposition instruction for DA Form 2408-13-1/2408-13-1-E in paragraph 2-9.D.(2) of DA Pamphlet 738-751.

INSPECTION AREAS. Figure 6 reflects the inspection areas of the AH-64D helicopter. Those areas are titled as shown. Figure 7 shows the location of access doors and panels which require removal at various phased maintenance inspections.

REPORTING ERRORS AND RECOMMENDING IMPROVEMENTS. You can help improve this manual. If you find any mistakes or if you know of a way to improve the procedures, please let us know. Mail your letter, DA Form 2028 (Recommended Changes to Publications and Blank Forms), located in the back of this manual direct to: Commander, U.S. Army Aviation and Missile Command, ATTN: AMSAM-MMC-MA-NP, Redstone Arsenal, AL, 35898-5000. A reply will be furnished to you. You may also provide DA Form 2028 information to AMCOM via e-mail, fax, or the World Wide Web. Our fax number is: DSN 788-6546 or Commercial 256-842-6546. Our e-mail address is: 2028@redstone.army.mil. Instructions for sending an electronic 2028 may be found at the back of this manual immediately preceding the hard copy 2028. For the World Wide Web use: https://amcom2028.redstone.army.mil.

РНА	SE NO1 P	HASE	D MAINTENANCE CHECKLIST			
L	Area Name and No. EFT FORWARD AVIONICS BAY AND MLG		Aircraft Serial No. 77–23259	Date 4 APR 81	Total Hrs. This Are	a
Inspect Phase Nos.	Inspection Requirements	Status	Faults and/or Remarks		Action Taken	Initial
ALL C 👡	1. MLG wheel for cracks, distortion, or corrosion. Hub for grease leakage. Tire for INDICATES COMBAT SITUATION INSPECTI	ON			Insp - OK	WCB
ALL C	MLG wheel brake for fluid leakage, cracked housing or corrosion. Check brake pucks for wear.	8	corrosion on bottom of brake housing WCB Leakage from housing inlet connection WCB (continued on Supplement	Inlet fit	ting tightened	WCB
2,4	 Search light for corrosion, loose or missing fasteners, and security. Lens for cracks or evidence of overheating. Wiring for loose connections. Chafing, deterioration, and security. 		THIS ITEM NOT APPLICABLE TO PHASE NO. 1 EXAMPLE Entries si	HEAVY SEPAR A BLO	N/A LINES ADDED TO ATE FAULTS WIT CK fictitious and are intended	HIN

Figure 1. Example of Phased Maintenance Checklist Title Sheet

РНА		a Name RANSM	and No. MISSION - 9	Aircraft Serial No. 77 – 23259	Date 4 APR 81
Inspect Phase Nos.	Inspection Requirements	Status	Faults and/or Remarks	Action Taken	Initial
2,4	5. Breathers cleaned.			INSP-OK	WCB
	Access L200, R200		EXM		
ALL	6. Lube oil and filters changed.		TWO PEO	Insp - OK PLE PERFORMED THIS	WCB GDL
	Access L200, R200		INSPECTION INITIALED	ON. BOTH HAVE	
ALL	7. Lube oil level sight gages for cleanliness, leakage, and security. Clean lenses.	B		Tightened SAME PERSON (WCB) D CORRECTED THE	WCB
	Access L200, R200		FAULT.		
2,4	Input shafts and couplings for cracks, dents, distortion, and corrosion.	X	No.2 shaft input Coupling diaphram cracked WCB	Coupling replaced QA INITIALS ON LAS	
	Access L200, R200, LN6, RN6		QA SIGN OFF	Insp-OK HaroldS.	
	Uncorrected Fault/ Discrepancy from DA form 2408-13 INSPECTION ITEM ADDED TO AVAIL SPACE ON A CHECKLIST PAGE.	ABLE	En	Rewired ON CORRECTED THIS tries shown above are fictitious and y to illustrate usage of form.	

Figure 2. Example of Phased Maintenance Checklist Continuation Sheet

	For use o	PHASED MAINTENANCE CH f this form, see TM 55-1510 series and TM 55-1520 series, the p			Readiness Command.	DATE 4 APR 8	
PHASE		AREA NAME AND NUMBER LEFT FWD AVIONICS BAY AND ML		AIRCRAFT SERIAL NO. 77-23259	TOTAL HOURS THIS ARE		<u>) I</u>
INSPECT AREA NO.	INSPECT ITEM NO	INSPECTION REQUIREMENTS	STATUS	FAULTS AND/OR REMARKS	ACTION TAKEN		INITIAL
3	9	(continued)	B	Brake puck worn, unserviceable wcb	Replaced		WCB
				UTSET VICEUBTE WCB			
		A SUPPLEMENTAL SHEET IS					
		USED WHEN SPACE IS NOT AVAILABLE ON CHECKLIST					
		PAGE FOR ALL FAULTS OR CORRECTIVE ACTION.					
		CORRECTIVE ACTION.					
				EXAME	<u>_</u>		
				-10	re		
				-XAM			
							
		Check work area for tool		DD REMINDER" s after completion of maintenance and Inspection.			

DA Form 4676-R 1 Dec 77

Figure 3. Example of Checklist Supplemental Sheet (Sheet 1 of 2)

F	For use o	f this form, see TM 55-1510 series and TM 55-1520 series, the	nconcent	T (SUPPLEMENTAL SHEET)		DATE
PHASE N		AREA NAME AND NUMBER	ргоропент а	AIRCRAFT SERIAL NO.	TOTAL HOURS THIS AREA	
INACEN	10.	ANEA NAME AND NOMBER		AINONAL LOENIAE NO.	TOTAL HOOKO THIO AKLA	
INSPECT I AREA NO.	INSPECT ITEM NO	INSPECTION REQUIREMENTS	STATUS	FAULTS AND/OR REMARKS	ACTION TAKEN	INITIAL
	l			 D REMINDER"		1
		Check work area for to		after completion of maintenance and Inspection.		

DA Form 4676-R 1 Dec 77

Figure 3. Example of Checklist Supplemental Sheet (Sheet 2 of 2)

A/C NO	PURPOSE OF	TEST	FLIC	SHT			DATE	
PILOT AND UNIT							TIME	
GROSS WEIGHT	C.G.	FAT		°C	PRESS AL	T	DENSI	TY ALT
SYMBOLS:	√ = \$	SATISF	AC	ГORY	× =	UNS	SATISFA	CTORY
PRIOR TO MT	F CHECKS		ST	ARTIN	IG ENGINES	S – P	ILOT	
INTERIOR CH	IECK-CPG			1. E	ENG 1 STAF	₹ T	· · ·	
INTERIOR CH	ECK-PILOT			a.	TIME TO ID	LE		SEC
BEFORE STARTI	NG APU-PILOT/	CPG		b.	IDLE SPEE	D		% N _G
1. ICS SYST	EM			C.	OIL PRESS	URE		PSI
2. FIRE DET	ECTORS			d.	TGT			°C
STARTING APU-I	PILOT			2.	ENG 2 STAF	₹T		
APU START				a.	TIME TO ID	LE		SEC
				b.	IDLE SPEE	D		% N _G
AFTER STARTING	G APU			C.	OIL PRESS	URE		PSI
1. EXT & IN	TR LIGHT			d.	TGT	_		°C
2. ECS SYS	TEM							
3. EGI				3. 1	N _P AND N _R	100%	101%	
4. BORESIG	HT NUMBERS							
5. IHADSS								
6. TADS SY	STEM CHECKS		EN	GINE	S RUNUP -	PILO	T	
	STEM CHECKS			1.	i "G" SPRIN	G		
8. FCR CHE	CKS L				TORQUE		%	%
9 WEAPON	S SYSTEMS			2. 1	ENGINE CH	OP C	IRCUIT	
10. FLIGHT C	CONTROLS CHE	CKS				,		
a. CONTRO	OL SWEEP				ENGINE OV	ERSI	PEED TI	EST
b. STABILA	TOR				ENG 1			
(1). MANU	AL			b.	ENG 2			
(2). AUTO								
c. BUCS C	HECKS B			4.	DEC/ECU L	OCK	TUC	
11. ANTI-ICE				a.	ENG 1			
a. PITOT H	EAT			b.	ENG 2			
b. CANOP	Y DEFOG							
12. POWER I	LEVERS			5.	FUEL SYST	EM		
13. ENG REI	NST/REPL CHE	CKS						

1-9

BEFORE TAXI CHECK	5. ATTITUDE HOLD
1. SYS PAGE 1 2	6. MANEUVERING FLIGHT
a. ENG OIL PRES	7. STABILATOR SYSTEM
b. NGB PRES	8. V _H CHECK
c. NGB TEMP	
d. XMSN PRES	ENGINE PERFORMANCE 701 701C
e. XMSN TEMP	9. MAX POWER CHECK
2. ENG PAGE 1 2	a. P. A. FT b. FAT °C
a. TORQUE	ENG 1 2
b. TGT	c. TGT °C °C
c. N _P	d. TORQUE % %
d. N _R	e. N _G % %
e. N _G	f. OIL PRESS PSI PSI
3. HIT CHECK	g. KTAS
	10. TGT LIM/CONTGCY PWR CK
TAXI CHECK	a. TORQUE % %
1. WHEEL BRAKES	b. TGT °C °C
2. INSTRUMENT & SYMBOLOGY	
	MISSION EQUIPT CHECKS
HOVER CHECKS	11. NAV COM EQUIPT CKS
1. INITIAL HOVER CHECK	a, ADF
2. NAV SYSTEMS CHECKS	b. NAV
3. HOVER MANEUVERING CHECK	c. XPNDR
4. FMC CHECK	d. COMM
5. VISIONIC SYSTEM CHECK	12. PNVS SYSTEMS CHECKS
6. HOVER BOX DRIFT	13. TADS SYSTEM CHECKS
FLIQUE OUEONO	14. FCR / BPS 15. WEAPON SYSTEM CHECKS
FLIGHT CHECKS 1. TAKEOFF	BEFORE LANDING
2. CRUISE	AFTER LANDING ENGINE SHUTDOWN
3. FUEL CONSUMPTION CHECK	
START STOP	1. PILOT
LBS PER HOUR	2. CPG
4. AUTOROTATION	3 EMERGENCY HYDRAULICS
a. PRESS ALT FT	4. UTIL ACC PRESS PSI
b. FAT °C	PEROPE LEAVING HELIOCOTES
c. N _R %	BEFORE LEAVING HELICOPTER
d. FUEL LB	I

REMARKS:		
SIGNATURE:		
I		

Figure 4. Example of Maintenance Test Flight Checksheet (Sheet 3 of 3)

		BLAI SERIAI	DE NO. 1 NUMBE	ER .				BLAD SERIAL	E NO. 2 NUMBE	R
ADJUSTMENT NUMBER	TAB	PITCH LINK ADJ	BALANCE	EFFECT		ADJUSTMENT NUMBER	TAB	PITCH LINK ADJ	BALANCE	EFFECT
1.						1.				
2.						2.				
3.						3.				
4.						4.				
5.						5.				
		BLAI SERIAI	DE NO. 3 _ NUMBE	EXA	\$	NP'		BLAD SERIAL	E NO. 4 NUMBE	R
ADJUSTMENT NUMBER	TAB	PITCH LINK ADJ	BALANCE	EFFECT		ADJUSTMENT NUMBER	TAB	PITCH LINK ADJ	BALANCE	EFFECT
1.						1.				
2.						2.				
3.						3.				
4.						4.				
5.						5.				
				RE	MA	ARKS				
									PILO1	SIGNATURE

Figure 5. Example of Rotor Smoothing Record

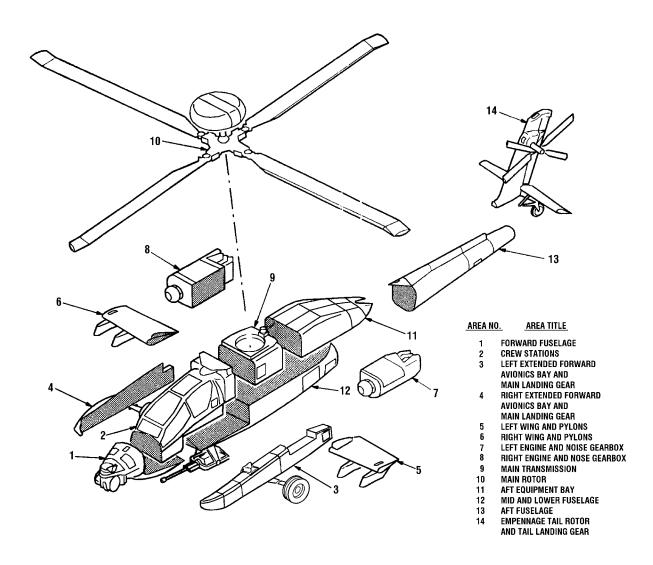


Figure 6. Inspection Area Diagram

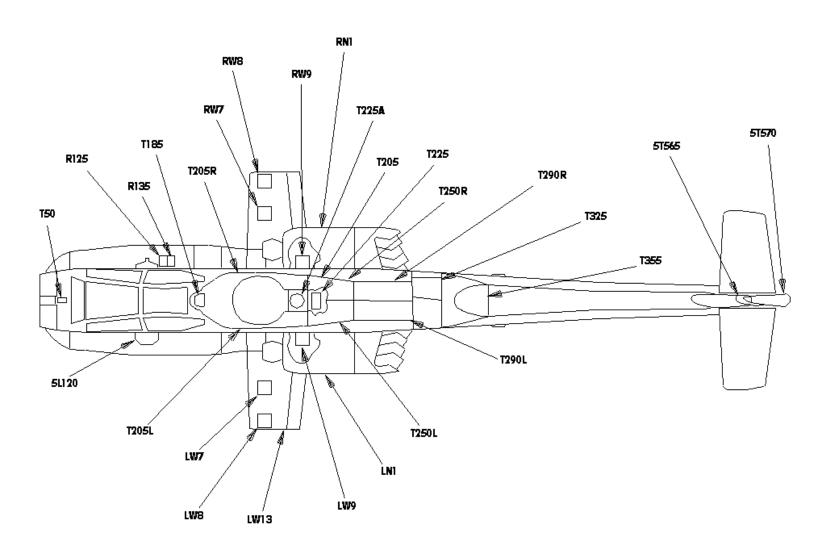


Figure 7. Inspection Access Provisions (Sheet 1 of 8)

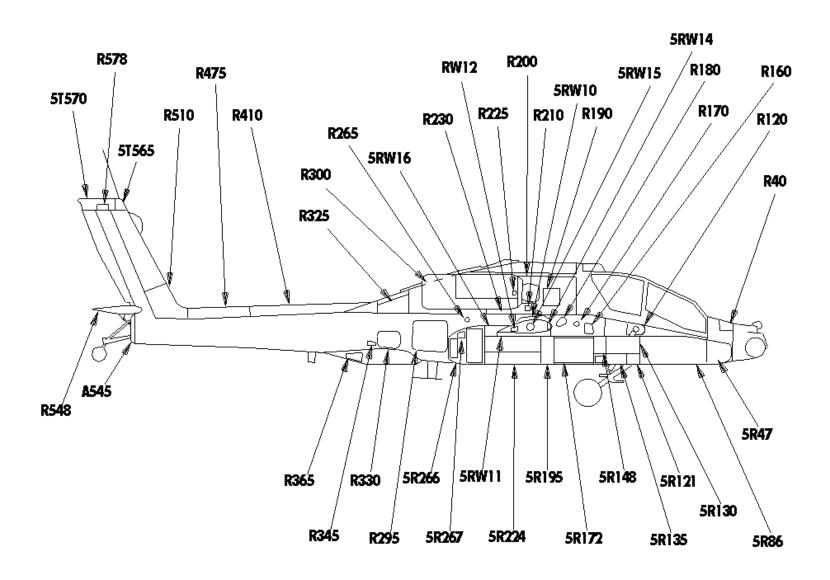
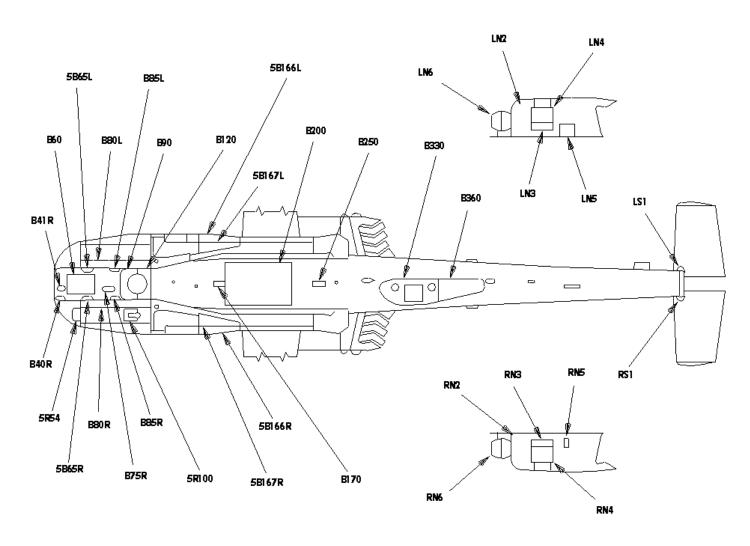


Figure 7. Inspection Access Provisions (Sheet 2 of 8)



773511_00.XD03A

Figure 7. Inspection Access Provisions (Sheet 3 of 8)

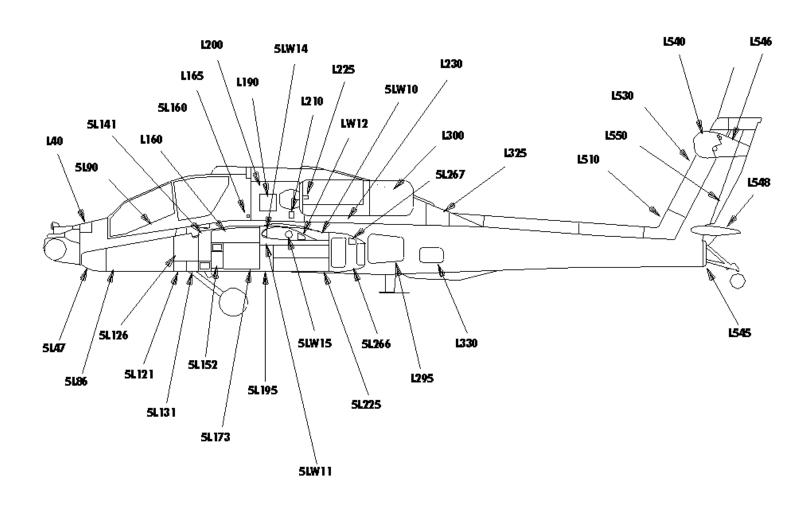
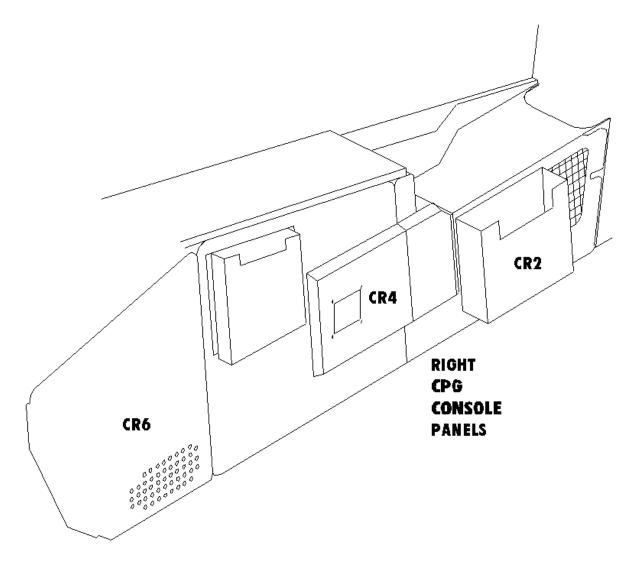


Figure 7. Inspection Access Provisions (Sheet 4 of 8)



77_00.XD19A

Figure 7. Inspection Access Provisions (Sheet 5 of 8)

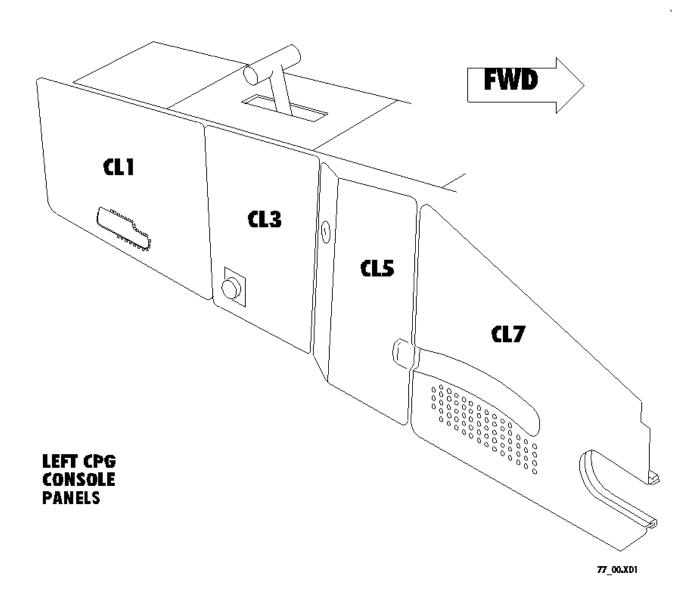


Figure 7. Inspection Access Provisions (Sheet 6 of 8)

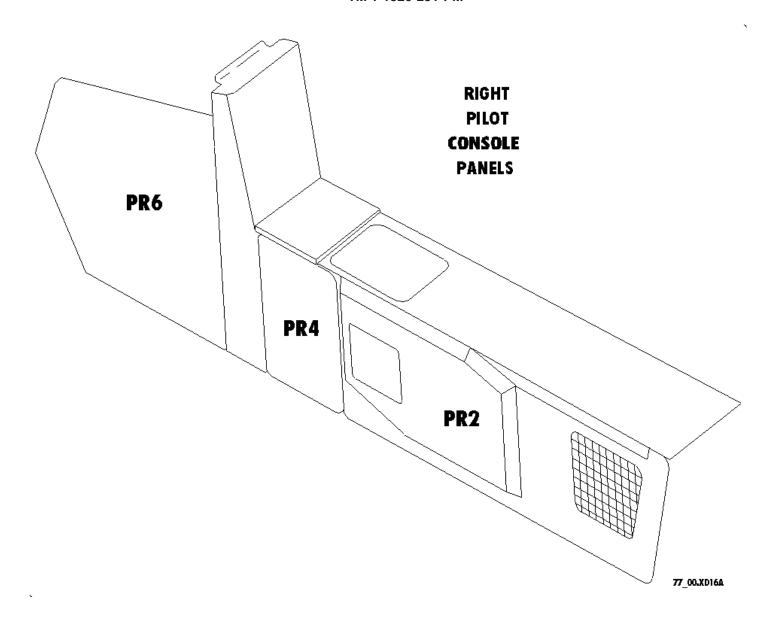
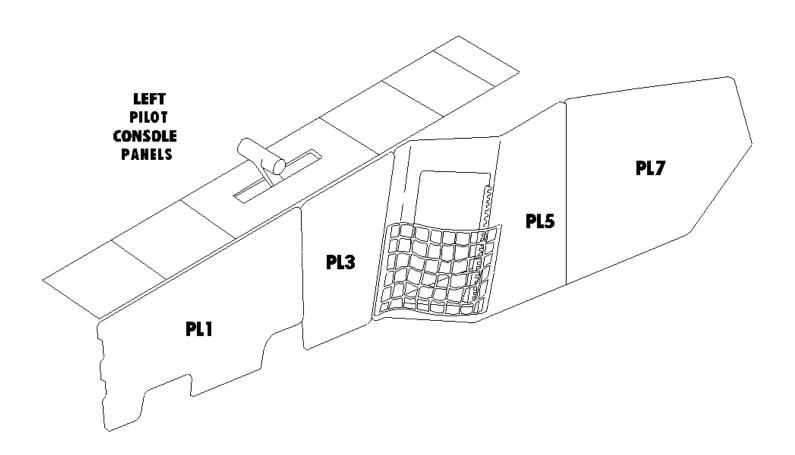


Figure 7. Inspection Access Provisions (Sheet 7 of 8)



77_00.XD15A

Figure 7. Inspection Access Provisions (Sheet 8 of 8)

Table I. Final Records Checklist

This checklist is provided to ensure that the indicated forms and records have been inspected for presence, completeness, legibility and accuracy prior to releasing the aircraft from a phased inspection. Verification of inspection will be indicated by placing the initials of the inspector in the appropriate initial block.

AIRCRAFT LOG BOOK	INITIAL	HISTORICAL RECORDS	INITIAL
DA FORM 2408		DA FORM 2408-5	
DA FORM 2408-12			
DA FORM 2408-13-1			
DA FORM 2408-14		DA FORM 2408-9	
DA FORM 2408-18		DA FORM 2408-15	
TM 1-1520-LONGBOW/APACHE		DA FORM 2408-16	
		DA FORM 2408-17	
TM 1-1520-251-MTF		DA FORM 2408-19	
LOCALLY REQUIRED FORMS		LOCALLY REQUIRED FORMS	
PRODUCTION CONTROL RECORDS	INITIAL	QUALITY CONTROL	INITIAL
FLOW CHART		TBO FILE	
STATUS BOARD		QA FILE	
WORK ORDER FILE		SERIAL NUMBER FILE	
MWO FILE			
WWO FILE		AOAP FILE	
CONFIGURATION CHART		AOAP FILE INVENTORY RECORDS	
CONFIGURATION CHART		INVENTORY RECORDS	
CONFIGURATION CHART 2405 LOG		INVENTORY RECORDS WEIGHT AND BALANCE	

Table II. Signature Sheet

Signature of Person Accomplishing Necessary Work	Initial
Signature of Person Accomplishing Necessary Work	Initial
Signature of Person Accomplishing Necessary Work	Initial
Signature of Person Accomplishing Necessary Work	Initial
Signature of Person Accomplishing Necessary Work	Initial
Signature of Person Accomplishing Necessary Work	Initial
	Initial
	Initial
Signature of Maintenance Supervisor	Initial
Signature of Technical Inspector	Initial
Signature of Maintenance Officer	Initial

Table III. Signature Sheet

Signature of Person Accomplishing Necessary Work	Initial
Signature of Person Accomplishing Necessary Work	Initial
Signature of Person Accomplishing Necessary Work	Initial
Signature of Person Accomplishing Necessary Work	Initial
Signature of Person Accomplishing Necessary Work	Initial
Signature of Person Accomplishing Necessary Work	Initial
	Initial
	Initial
Signature of Maintenance Supervisor	Initial
Signature of Technical Inspector	Initial
Signature of Maintenance Officer	Initial

SECTION II. INSPECTION CHECKLIST

WARNING

ACCIDENTAL ACTUATION OF HELICOPTER POWER PLANT, HYDRAULIC SYSTEM, CANOPY JETTISON SYSTEM, OR FIRING OF ARMAMENT OR STORES JETTISON BALLISTICS MAY CAUSE SEVERE INJURY OR DEATH. BEFORE STARTING INSPECTION, HELICOPTER SAFETY CHECK MUST BE PERFORMED (TM 1-1520-LONGBOW/APACHE) AND ALL ARMAMENT MUST BE SAFETIED, DEACTIVATED, AND CLEARED (TM 9-1090-208-23 AND TM 9-1427-475-23).

NOTE

Prior to the start of the phased maintenance inspection, it is recommended that a pre-inspection Maintenance Test Flight (MTF) be conducted. Accomplishment of the MTF shall be determined by the unit maintenance officer. The pre-inspection MTF should be conducted by a maintenance test pilot following a review of the aircraft forms and records and a briefing from the crew of the helicopter. The MTF is recommended to assess the helicopter performance and identify deficiencies that should be corrected while the helicopter is undergoing phased maintenance inspections.

РНА	PHASE NO PHASED MAINTENANCE CHECKLIST							
		Area Name and No. GENERAL	Aircraft Serial No.		Date	Total Hrs. This Area	l	
Inspect Phase Nos.		Inspection Requirements	Status	Faults and/or Remarks	Action Taken		Initial	
ALL C	1.	Prior to inspection, check forms and records for record deficiencies.						
ALL	1a.	Clean engines.						
	2.	Deleted.						
	3.	Deleted.						
ALL C	4.	Take oil samples (nose gearboxes, main transmission, and APU) within 30 minutes of engine shutdown.						
ALL C	5.	Fuel tanks will be fully serviiced prior to start of phased inspection. If maintenance is to be accomplished which requires defueling, this item may be deferred until after such maintenance is completed.						
ALL	6.	Depanel aircraft.						

[&]quot;FOD REMINDER" Check work area for tools and parts after completion of maintenance and inspections.

2-2 C1

РНА	PHASE NO PHASED MAINTENANCE CHECKLIST							
Area Name and No. GENERAL		Aircraft Serial No.		Date	Total Hrs. This Area			
Inspect Phase Nos.	Inspection Requirements	Status	Faults and/or Remarks	Action Taken		Initial		
ALL	7. Remove gun turret.							
ALL	8. Remove ammo magazine.							
ALL	9. Wash aircraft.							
ALL	10. Remove main rotor blades.							
	11. Deleted.							
ALL	12. Remove pilot seat.							
ALL	13. Remove CPG seat.							
ALL	14. Remove pylon ejector cartridges and stow IAW unit SOP.							

TM 1-1520-251-PM

PHA	PHASE NO PHASED MAINTENANCE CHECKLIST							
	Area Name and No. FORWARD FUSELAGE – 1		Aircraft Serial No.		Date	Total Hrs. This Area	_	
Inspect Phase Nos.		Inspection Requirements	Status	Faults and/or Remarks	Action Taken		Initial	
ALL	1.	(WARNING Do not attempt to move jettison handle).						
		Exterior skin for holes, cracks, dents, corrosion, loose or working rivets, and loose or missing attaching hardware.						
ALL	2.	Access panels, fairings, and doors for deformations, cracks, corrosion, loose or missing hardware. Door and hinges for damage, binding, and security. Latches for security and proper operation. Check for worn or deteriorated seals and copper coat.						
ALL	3.	Interior components for mounting security and loose or missing hardware.						
ALL	4.	Interior structure for cracks, corrosion, loose or working rivets, and loose or missing hardware.						

РНА	PHASE NO PHASED MAINTENANCE CHECKLIST							
Area Name and No. FORWARD FUSELAGE – 1		Aircraft Serial No.		Date	Total Hrs. This Area			
Inspect Phase Nos.		Inspection Requirements	Status	Faults and/or Remarks	Action Taken		Initial	
ALL	5.	Avionics Equipment for proper storage, external damage, and loose connectors. Wiring harness for chafing and deterioration.						
ALL	6.	CPG brake master cylinders for leakage, cracks, and loose or missing hardware. None allowed, if found replace with undamaged like item. Hydraulic lines for leakage, chafing and connection security.						
ALL C	7.	Flight control rods for dents, cracks, corrosion, security and evidence of interference. Rod ends for worn or seized bearings.						
ALL C	8.	Flight control bellcranks for cracks, corrosion, security and evidence of interference. Check pivot bearings for wear.						

PHA	PHASE NO PHASED MAINTENANCE CHECKLIST							
	Area Name and No. FORWARD FUSELAGE – 1		Aircraft Serial No.		Date	Total Hrs. This Area	1	
Inspect Phase Nos.		Inspection Requirements	Status	Faults and/or Remarks	Action Taken		Initial	
ALL C	9.	CPG Decoupler (ARDD) units for cracks, corrosion, security, and evidence of interference. Check for looseness and lost motion.						
ALL C	10.	LVDT(S) for cracks, corrosion, and mounting security. Wiring harnesses for loose connections, chafing, or deterioration and evidence of interference.						
ALL C	11.	Exterior canopy jettison components for cut or broken transfer tube, bulged, chafed or swollen union, loose or missing hardware and safety wire torn or missing streamer.						
ALL	12.	Canopy emergency release drain hose and outlet for breaks.						

TM 1-1520-251-PM

PHA	SE	NO	HASED	MAINTENANCE CHECKLIST			
	ARN	Area Name and No. MAMENT FORWARD FUSELAGE – 1	Aircraft Serial No.		Date	Total Hrs. This Ar	ea
Inspect Phase Nos.		Inspection Requirements	Status	Faults and/or Remarks	A	ction Taken	Initial
ALL	1.	Aircraft interface assembly and fairings for cracks, distortion, missing fastners and damage.					
ALL	Aircraft Interface Assembly wiring harnesses for chafed or broken wires and loose, bent, burned or broken pins or sockets.						
ALL	3.	. Glass gaskets on both dayside and nightside sensor assemblies for damage.					
ALL	4.	AIA bonding strap assembly for loose, damaged and frayed wires. Check bonding strap assembly mounting hardware for security and damage.					
ALL	Area around lower support laser shield for cracked or peeling paint.						
ALL	L 6. Boresight assembly optics for contamination.						

[&]quot;FOD REMINDER" Check work area for tools and parts after completion of maintenance and inspection.

PHA	PHASE NO PHASED MAINTENANCE CHECKLIST									
	Area Name and No. ARMAMENT FORWARD FUSELAGE – 1		Aircraft Serial No.	Date	Total Hrs. This Area	l				
Inspect Phase Nos.	Inspection Requirements	Status	Faults and/or Remarks		Action Taken	Initial				
ALL	7. Area around boresight assembly for cracked, broken, or peeling paint.									
ALL	ALL 8. ECA for external damage.									
ALL	Remove and replace ECS air filter assembly.									
ALL	ECS molded hose for cracks, cuts, deterioration, and loose fit.									
ALL	ALL 11. Check external surfaces of night sensor shroud for dents, cracks, punctures, and window for cracks or chips.									
ALL	ALL 12. Remove night sensor shroud assembly. Inspect inner surface of window for contamination, pitting, nicks and scratches.									

PHA	SE NO P	HASED N	MAINTENANCE CHECKLIST			
	Area Name and No. ARMAMENT FORWARD FUSELAGE – 1		Aircraft Serial No.	Date	Total Hrs. This Area	1
Inspect Phase Nos.	Inspection Requirements	Status	Faults and/or Remarks	A	ction Taken	Initial
ALL	13. Internal painted surfaces for scratched, chipped, or peeled paint. Check for any damage to components. Relate scratches and chips to possible points of contacts.					
ALL	14. NSA harnesses for chafed or broken wires and loose or corroded connections. Anti-ice CCA A1 and connector P1 for loose mounting hardware. Window temperature sensor RT1 and thermostat S2 for bonding separation.					
ALL	15. NSA optics for contamination, pitting, nicks, scratches and peeling of coating.					
ALL	.L 16. External surfaces of day shroud for dents, cracks or punctures and window for chips or cracks.					
ALL	LL 17. Remove day sensor shroud assembly. Inspect innner surfaces of window for contamination, pitting, nicks, scratches and peeling of coating.					

[&]quot;FOD REMINDER" Check work area for tools and parts after completion of maintenance and inspection.

TM 1-1520-251-PM

РНА	SE NO F	PHASED I	MAINTENANCE CHECKLIST			
	Area Name and No. ARMAMENT FORWARD FUSELAGE – 1	Aircraft Serial No.		Date	Total Hrs. This Area	1
Inspect Phase Nos.	Inspection Requirements	Status	Faults and/or Remarks	А	action Taken	Initial
ALL	L 18. Internal painted surfaces for scratched, chipped, or peeled paint. Check for any damage to components. Relate scratches and chips to possible points of contact.					
ALL	19. Harnesses for chafed or broken wires and loose or corroded connections. Anti-ice CCA A1 and connector P1 for loose mounting hardware. Window temperature sensor RT1 and thermostat S1 for bonding separation.					
ALL	DSA optics for contamination, nicks, scratches, and peeling of optical coating.					
ALL	ALL 21. PNVS shroud assembly external surfaces for dents, cracks, chips, punctures, scratches, peeled paint, and window for chips or cracks.					
ALL	22. Remove PNVS shroud. Inner surface of window for contamination, pitting, nicks, scratches, chips, and peeled paint.					

TM 1-1520-251-PM

PHA	SE	NO P	HASED N	MAINTENANCE CHECKLIST			
	ARN	Area Name and No. MAMENT FORWARD FUSELAGE – 1	Aircraft Serial No.		Date	Total Hrs. This Are	ea
Inspect Phase Nos.		Inspection Requirements	Status	Faults and/or Remarks	A	ction Taken	Initial
ALL	chipped, or peeled paint. Check for any damage to components. Relate scratches and chips to possible points of contact.						
ALL							
ALL	25.	Electronic shielding gasket for bonding separation, tears, cracks, cuts, and deterioration.					
ALL	ALL 26. PNVS optics for contamination, pitting, nicks, scratching, and peeling of coating.						
ALL	27. Wire harnesses for chafed or broken wires, loose or corroded connections and loose connectors.						

РНА	PHASE NO PHASED MAINTENANCE CHECKLIST									
	Area Name and No. ARMAMENT FORWARD FUSELAGE – 1		Aircraft Serial No.	Date	Total Hrs. This Area	ı				
Inspect Phase Nos.	Inspection Requirements	Status	Faults and/or Remarks	Action Taken		Initial				
ALL	28. Check for loose, stripped or missing mounting hardware on PNVS turret.									
ALL	ALL 29. Main fan, mux power supply fan, and two ventilated CCA housings for buildup of dust.									
ALL	L 30. Access cover seal for any tears, cracks, cuts, deterioration, and loose fit.									
ALL	31. Remove PNVS turret assembly.									
ALL	32. PNVS turret seal for any cracks, tears, bonding separation, or deterioration. Gears for any cracked, broken, or missing teeth. Telfon limit stop pads for any damage.									
ALL	ALL 33. PNVS turret captive mounting screws, stripped, damaged or missing screws.									
ALL	34. Broken or bent switch actuators and loose limit switches.									

PHA	SE	NO P	HASED	MAINTENANCE CHECKLIST			
	ARN	Area Name and No. MAMENT FORWARD FUSELAGE – 1		Aircraft Serial No.	Date	Total Hrs. This Area	1
Inspect Phase Nos.		Inspection Requirements	Status	Faults and/or Remarks	,	Action Taken	Initial
ALL	35. Move TADS turret to 90 degrees. Check external painted surfaces of PECA for scratched, chipped, or peeled paint.						
ALL	36.	Check PECA warning plate decal for peeling edges. Check for extensive wear, pitting, legibility, etc.					
ALL	ALL 37. ALQ 136 antenna and associated wiring for mounting security, external damage, and loose connectors. Wiring harnesses and clamps for chafing security, and deterioration.						

TM 1-1520-251-PM

РНА	SE	NO P	HASE	D MAINTENANCE CHECKLIST			
		Area Name and No. CREW STATIONS – 2		Aircraft Serial No.	Date	Total Hrs. This Area	l
Inspect Phase Nos.		Inspection Requirements	Status	Faults and/or Remarks		Action Taken	Initial
ALL	1.	WARNING: DO NOT MOVE HEIGHT					
/\	l ''	ADJUSTMENT LEVER UNLESS SEAT IS					
		OCCUPIED. IF INJURY OCCURS, SEEK					
		MEDICAL AID.					
		Exterior skin for holes, cracks, dents,					
		corrosion, loose or working rivets, and loose					
		or missing attaching hardware. Handholds and steps for damage, structural integrity,					
		and mounting security.					
ALL	2.	Access panels, fairings, and doors for					
		deformation, cracks, corrosion, loose or					
		working rivets, and loose or missing hardware. Door hinges for damage, binding					
		and security. Latches for security and					
		proper operation. Check for worn or					
		deteriorated seals.					
ALL	3.	Interior components for mounting security					
/\	0.	and loose or missing hardware.					
		and loose of missing nardware.					
ALL	4.	Interior structure for cracks, corrosion, loose					
		or working rivets, and loose or missing					
		hardware.	<u> </u>				

РНА	SE	NO		a Name N STAT	and No. IONS – 2		Aircraft Serial No.	Dat	е
Inspect Phase Nos.		Inspection Red	quirements	Status	Faults and/or Rema	rks Action Taken			Initial
2,4	5.	Pilot door for seal deterioration and proper fit. Hinges for cracks and corrosion. Door mechanism for damage and loose door strut cable. Cracked or broken striker plate. Door strut for corrosion and loose or worn bearings. Remove pilot floor panels for access. Pilot magnetic brake trim and feel spring units for cracks, corrosion, and mounting security. Check for looseness or lost motion. Wiring harnesses for loose connections, chafing, or deteriorations. Pilot flight control linkage for cracks, corrosion, and security.							
2,4 C	6.								
2,4 C	7.								
ALL	8.	High powered switch mounting security. W proper connections, of cleanliness. Circuit b and damage. Check readability.	iring harnesses for chafing, and reakers for looseness						

[&]quot;FOD REMINDER" Check work area for tools and parts after completion of maintenance and inspections.

РНА	SE	NO		a Name a W STAT	and No. IONS – 2		Aircraft Serial No.	Date	е
Inspect Phase Nos.		Inspection Red	quirements	Status	Faults and/or Rema	rks	Action Taken		Initial
ALL	9.	, ,							
		loose or missing hardware. Loose, cracked,							
		or broken seat attachment fittings.							
ALL	40	Dilet eact for ereals	distantian and sociulty						
ALL	10.	Upholstery and cush	distortion, and security.						
		cleanliness.	ions for todis and						
ALL	11.								
			g, and cleanliness. Strap fittings for						
		corrosion and securit	ty.						
ALL	12.	Pilot seat belt buckle	for proper operation.						
,			io. propor operanom						
ALL	13.								
		release operation. Re	eleased harness straps						
		for free extension an	d reel-in operation.						
ALL	14.	Pilot seat armor swin	ng plate for mounting						
,,,,,	1-7.	security, lockup, lock	release, and side						
		swing operation.	,						

РНА	SE N	NO		a Name a	and No. IONS – 2		Aircraft Serial No.	Date	е
Inspect Phase Nos.		Inspection Rec	quirements	Status	Faults and/or Remark	ks	Action Taken		Initial
ALL		5. Pilot instrument panels, MPD's, and consoles for loose or missing fasteners. MPD's, UFD, compass, control knobs and switches for damage or looseness. Lenses for transparency and cleanliness. Check all markings for readability.							
ALL		Pilot cyclic stick for so bushings. Grip switch looseness. Check all readability. Base wirin connections, chafing, Control linkage for da evidence of interferen	mes for damage or markings for ng harness for loose , and deterioration. amage, looseness, and						
ALL		Pilot collective stick for security and worn bushings. Grip switches for damage or looseness. Check all markings for readability. Base wiring harness for loose connections, chafing, and deterioration. Control linkage for damage, looseness, and evidence of interference.							
ALL	18. Pilot directional pedals for damage and security. Supports for cracks, bends, or corrosion. Rod and rod ends for cracks, bending, corrosion, worn or seized bearings, and loose or missing hardware.								

PHA	SE	NO		a Name a W STAT	and No. IONS – 2		Aircraft Serial No.	Dat	е
Inspect Phase Nos.		Inspection Red	quirements	Status	Faults and/or Rema	rks	Action Taken		Initial
	40 8"								
2,4	19.	Pilot engine power co	ontrols for bent,						
С		cracked or broken cable supports, and brackets. Rods and rod ends for worn or							
		seized bearings and loose or missing							
		hardware. Remove pilot's left console side							
		panels for access.							
2,4	20.	Pilot wiring harnesses for loose and missing attaching hardware, chafed wires, loose							
С		connections, and bro							
		Brackets for damage							
2,4	21.		ls for damage, security						
		and loose or missing	hardware.						
ALL	22	Dilat canany inttinan	components for out or						
ALL	22.	broken detonation co	components for cut or						
С		and loose or missing							
		and loose or missing hardware.							
ALL	23.	Doghouse fairing for	damage. Check for						
		security and integrity	of mounted jammer						
		antenna, IFF antenna	a, and de-ice probe.						

PHA	SE	NO		a Name a N STATI	and No. IONS – 2		Aircraft Serial No.	Date	е
Inspect Phase Nos.		Inspection Red	quirements	Status	Faults and/or Remar	rks	Action Taken		Initial
ALL	24.								
			elease harness straps						
		for free extension and	d reel-in operation.						
2.4	25	CDC door for oool do	stariaration and proper						
2,4	25.	fit. Hinges for cracks							
С			age and loose door strut						
			roken or cracked. Door						
			trut for damage and worn or seized						
		bearings.	irings.						
ALL	26.								
		and side swing opera	lockup for lock release						
		and side swing opera	ation.						
ALL	27.	CPG seat for cracks,	distortion and						
ALL	21.	•	and cushions for tears						
		and cleanliness.	and oddinono for todio						
ALL	28.	CPG seat height adju	ustment track for height						
		limiter.							
ALL	29.		arness straps for cuts,						
		fraying, and cleanline							
		corrosion and securit	ty.						
		correction and security.							

РНА	SEI	NO		a Name N STAT	and No. IONS – 2		Aircraft Serial No.	Dat	е
Inspect Phase Nos.		Inspection Red	quirements	Status	Faults and/or Rema	arks	Action Taken		Initial
ALL	30.	CPG instrument panel, MPD's, and consoles for loose or missing fastners. MPD's, UFD, control knobs and switches for damage or looseness. Lenses for transparency and cleanliness. Check all markings for readability.							
ALL	31.	CPG seat belt buckle for proper operation.							
ALL C	32.	 CPG cyclic stick for security and worn bushings. Grip switches for damage or looseness. Check all markings for readability. Base wiring harness for loose connections, chafing, and deterioration. Control linkage for damage, looseness, and evidence of interference. CPG collective stick for security and worn bushings. Grip switches for damage or looseness. Check all markings for readability. Base wiring harness for loose connections, chafing, and deterioration. Control linkage for damage, looseness, and evidence of interference. 							
ALL C	33.								

РНА	SE	NO		a Name a W STATI	and No. IONS – 2		Aircraft Serial No.	Date	е
Inspect Phase Nos.		Inspection Red	quirements	Status	Faults and/or Rema	rks	Action Taken		Initial
ALL C	34.	34. CPG directional pedals for damage and security. Supports for cracks, bends, and corrosion. Rod and rod ends for cracks, bending, corrosion, worn or seized bearings and loose or missing attachment hardware.							
2,4 C	35.	5. CPG engine power controls for bent, cracked, or broken cable supports and brackets. Rods and rod ends for worn or seized bearings and loose or missing hardware. Remove CPG left console side panel for access.							
2,4 C	36.	Remove CPG floor p flight control linkage and security.	eanels for access. CPG for cracks, corrosion						
2,4 C	37.	 CPG wiring harnesses for loose or missing hardware, chafed wires, loose connections, and broken tie-wraps. Brackets for damage and corrosion. 							
ALL	38. CPG SSU mount pads and struts for damage, security, and loose or missing hardware.								

РНА	SE N	NO		a Name a W STATI	and No. ONS – 2		Aircraft Serial No.	Date	е
Inspect Phase Nos.		Inspection Red	quirements	Status	Faults and/or Rema	rks	Action Taken		Initial
2,4 C	39.	broken detonation cords, bulged unions, and loose or missing hardware. Pilot brake master cylinders for leakage, cracks, and loose or missing hardware. Hydraulic lines for leakage, chafing, and connection security.							
ALL	40.								
ALL	41.								

PHA	PHASE NO PHASED MAINTENANCE CHECKLIST												
	Α	Area Name and No. ARMAMENT CREW STATIONS – 2		Aircraft Serial No.	Date	Total Hrs. This Area	1						
Inspect Phase Nos.		Inspection Requirements	Status	Faults and/or Remarks	Action Taken		Initial						
ALL	1.	Remove pilot HDU from crewstation.											
ALL	2.	Visually inspect pilot SSU, HDU, and BRU optical lenses for contamination, clean as required.											
ALL	3.	Inspect pilot BRU housing for damage. None allowed.											
2,4	4.	Inspect pilot BRU wiring harness, connector, and receptacle for loose, broken, or missing pins or damage. None allowed.											
ALL	5.	Inspect pilot display adjust panel (DAP) for cracks, corrosion, damaged connector receptables, and loose or missing hardware. Cracks: None allowed. If found, replace with undamaged like item.											

РНА	SE	NO	_	a Name ε ΓCREW	and No. / STATIONS – 2		Aircraft Serial No.	Date	е
Inspect Phase Nos.		Inspection Red	quirements	Status	Faults and/or Rema	rks	Action Taken		Initial
ALL	6.	Pilot Sensor Surveyir components for cracl connector receptacle allowed, if found replike item.	ks, and damaged . Damage: none						
ALL	7.	Visually inspect CPG SSU, HDU, and BRU optical lenses for contamination, clean as required.							
ALL	8.	Remove CPG HDU from crewstation.							
ALL	9.	Inspect CPG BRU ho connectors and recep broken or missing pir	ptacles for loose,						
ALL	10.	 Inspect CPG display adjust panel (DAP) for cracks, corrosion, damaged connector receptacles and loose or missing hardware. Cracks; none allowed. If found, replace with undamaged like item. 							
ALL	CPG Sensor Surveying Unit (SSU) component for cracks, and damaged connector receptacle.								

РНА	SE	NO		a Name a ΓCREW	and No. 'STATIONS – 2		Aircraft Serial No.	Date	е
Inspect Phase Nos.		Inspection Red	quirements	Status	Faults and/or Rema	rks	Action Taken		Initial
ALL	12.	NOTE: THE FOLLOWING PROCEDURES ARE APPLICABLE TO BOTH THE PILOT AND CPG HDU. Remove combiner lens from HDU and clean attaching parts.							
ALL	13.	Inspect the HDU housing, two screws, flat spring, and thrust plate for negligible or replacement damage.							
ALL	14.	Inspect HDU electrical connectors for broken, bent, loose or missing pins and foreign debris.							
ALL	15.		s for chips, cracks, and ose, broken or missing						
ALL	16.	Inspect CRT cable fo	or twisting or binding.						
ALL	17.	. HDU for damage, cracks, corrosion, loose or missing hardware, or connectors.							
ALL	18.	 Rotating ring, rotating segment, flat spring, and two screws for negligible or replacement damage. 							

РНА				a Name and No. T CREW STATIONS – 2		Aircraft Serial No.		Date	е
Inspect Phase Nos.		Inspection Requirements		Status	Faults and/or Rema	rks	Action Taken		Initial
ALL	19.	19. Perform HDU alignment verification.							
ALL	20.	 ORT painted surfaces for cracked, broken, or peeling paint, optics for contamination, pitting, nicks, scratching and ORT humidity indicator. 							
ALL	21.	. Control panel filter assembly for unrestricted movement and security in the up position.							
ALL	22.	 Check for deteriorated or missing snubber pads. 							

РНА	PHASE NO PHASED MAINTENANCE CHECKLIST												
		Area Name and No. LEFT EFAB AND MLG - 3		Aircraft Serial No.	Date	Total Hrs. This Area	3						
Inspect Phase Nos.		Inspection Requirements	Status	Faults and/or Remarks		Action Taken	Initial						
ALL	1.	Exterior skin for holes, cracks, dents, corrosion, loose or working rivets, and loose or missing hardware. Handholds and steps for damage, structural integrity, and mounting security.											
ALL	2.	Access panels, fairings, and doors for deformation, cracks, corrosion, loose or working rivets, and loose or missing hardware. Door hinges for damage, binding, and security. Latches for security and proper operation. Check for worn or deteriorated seals and copper cote.											
ALL	3.	Interior structure for cracks, corrosion, loose or working rivets, and loose or missing hardware.											
ALL C	4.	ECS compressor, condenser, evaporator and tubes for leaking and integrity. All ECS system components for proper charge, signs of leakage or overheating and proper installation.											

"FOD REMINDER"

PHA	SE	NO		a Name FAB AN	and No. ID MLG – 3		Aircraft Serial No.	Dat	е
Inspect Phase Nos.		Inspection Re	quirements	Status	Faults and/or Rema	ırks	Action Taken		Initial
ALL C	5.	Mounting racks for di missing hardware. C distortion, proper fit, leakage.	ooling ducts for cracks,						
ALL C	6.	Avionics bay for clea	nliness and distortion.						
ALL C	7.	(Remove shock strut) Inspect main landing gear shock strut mount from the base of the mount to the end of the shaft. Inspect for pits, grooves and scratches. For cracks use flourescent penetrant. Check for distortion, or looseness, loose or missing rivets or hardware. (Remove shock strut).							
ALL C	8.	Inspect main landing gear shock strut and bearings for damage.							
2,4 C	9.	Main landing gear trailing arm for cracks, distortion, and corrosion. Cross tube end pivot boss for security, damaged or worn bearing, loose or missing hardware. Hydraulic brake line for leakage, chafing, dents, corrosion, and clamping security.							

РНА	PHASE NO LEFT			a Name FAB AN	and No. ID MLG – 3		Aircraft Serial No.	Dat	e
Inspect Phase Nos.		Inspection Red	quirements	Status	Faults and/or Rema	arks	Action Taken		Initial
ALL C	10.	Main landing gear en cracks, distortion, fra elongated bolt holes.							
2,4 C	11.								
2,4 C	12. Main landing gear wheel keys and key slots for wear and damage.								
2,4	13. Squat switch for deformation, damaged insulation, loose connections, and mounting security. Wiring harness for chafing, security, and deterioration.		nections, and mounting ess for chafing,						
ALL C	14. Pilot collective bellcrank for chafing and interference with forward fuel cell. Check for structural deterioration and security. (Access L160)								

PHA	SE NO	0	Area Name and No. LEFT EFAB AND MLG – 3				Aircraft Serial No.	Date	е
Inspect Phase Nos.		Inspection Requirements			Faults and/or Rema	rks	Action Taken		Initial
ALL	fo	5. Pilot collective push-pull rods and rod ends for cranks, corrosion, bending, worn and seized bearings, loose or missing hardware, and evidence of interference.							
С									

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РНА	SE	NO P	HASEI	D MAINTENANCE CHECKLIST			
	AF	Area Name and No. RMAMENT LEFT EFAB AND MLG – 3		Aircraft Serial No.	Date	Total Hrs. This Area	ı
Inspect Phase Nos.		Inspection Requirements	Status	Faults and/or Remarks		Action Taken	Initial
ALL	1.	Connectors J1 through J9 for corrosion, damage, distortions, and loose mounting connections.					
ALL	2.	. Stow assembly mounting points for cracks and corrosion.					
ALL	3.	TEU assembly for loose, missing, or stripped mounting hardware. Remove TEU.					
ALL	4.	Air holes in air flow adapters for build-up of dust.					
ALL	Remove TPS. Check TPS for loose, missing, and stripped mounting hardware.						
ALL	6.	Connectors J1 and J2 for corrosion, loose connections and loose, bent, or broken pins and sockets.					

РНА	SE	NO		a Name a EFT EF	and No. FAB AND MLG – 3		Aircraft Serial No.	Dat	е
Inspect Phase Nos.		Inspection Red	quirements	Status	Faults and/or Rema	arks	Action Taken		Initial
ALL	7.	Fuses F1, F2, and F3 incorrect fuses.	3 for missing, blown, or						
ALL	8.	Check air holes in air flow adapters on rear of TPS for build-up of dust.							
ALL	9.	. Connectors J2 and J3 for corrosion, loose connections and for loose, bent, or broken pins and sockets.							
ALL	10.	LEU for loose, missir mounting hardware.							
ALL	11.	11. Avionics equipment for mounting security, external damage, and loose connectors. Wiring harnesses and clamps for chafing, security, and deterioration.							
ALL	12.	Wire harnesses for chafing, broken wires, and loose, bent, broken pins and sockets.							

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PHA	SE	NO P	HASED M	AINTENANCE CHECKLIST			
		Area Name and No. RIGHT EFAB AND MLG – 4		Aircraft Serial No.	Date	Total Hrs. This Are	a
Inspect Phase Nos.		Inspection Requirements	Status	Faults and/or Remarks	A	Action Taken	Initial
ALL	1.	Exterior skin for holes, cracks, dents, corrosion, loose or working rivets, and loose or missing hardware. Step for damage, structural integrity, and mounting security.					
ALL	2.	Access panels, fairings, and doors for deformation cracks, corrosion, loose or working rivets, and loose or missing hardware. Door hinges for damage, binding, and security. Latches for security and proper operation. Seals and copper cote for wear or deterioration.					
ALL	3.	Interior structure for cracks, corrosion, loose or working rivets, and loose or missing hardware.					
ALL	4.	ECS compressor, condenser, evaporators, and tubes for leaking and integrity. All ECS system components for proper charge. Signs of leakage or over heating and proper installation.					

[&]quot;FOD REMINDER" Check work area for tools and parts after completion of maintenance and inspections.

РНА	SE	NO		a Name FAB Al	and No. ND MLG – 4		Aircraft Serial No.	Dat	е
Inspect Phase Nos.		Inspection Red	quirements	Status	Faults and/or Rema	arks Action Taken			Initial
ALL	5.	Mounting rack for distortion and loose or missing hardware. Cooling ducts for cracks, distortion, proper fit, and evidence of leakage.							
ALL	6.	Avionics bay for cleanliness and distortion.							
C C	7.	(Remove shock strut) Inspect main landing gear shock strut mount from the base of the mount to the end of the shaft. Inspect for pits, grooves and scratches. For cracks use flourescent penetrant. Check for distortion, and loose or missing hardware. (Remove shock strut).							
ALL C	8.	Inspect main landing gear shock strut and bearings for damage.							
2,4 C	9.	Main landing gear tradistortion, and corrospivot boss for security bearing, loose or mis Hydraulic brake line fidents, corrosion, and	sion. Cross tube end y, damaged or worn sing hardware. for leakage, chafing,						

[&]quot;FOD REMINDER" Check work area for tools and parts after completion of maintenance and inspections.

РНА	SE NO		Area Name and No. RIGHT EFAB AND MLG – 4			Aircraft Serial No. Da		
Inspect Phase Nos.	Inspection Re	quirements	Status	Faults and/or Rema	arks	Action Taken		Initial
ALL C	10. Main landing gear end cap on cross tube for cracks, distortion, fractured weld, and elongated bolt holes.11. Repack main landing gear wheel bearings.							
2,4 C								
2,4 C		Main landing gear wheel keys and key slots for damage and wear.						
ALL C	13. Extend searchlight. (loose or missing hard Lens for cracks and overheating. Wiring f chafing, deterioration	dware, and security. evidence of for loose connections,						

[&]quot;FOD REMINDER" Check work area for tools and parts after completion of maintenance and inspections.

PHA	SE	NO P	HASE	MAINTENANCE CHECKLIST			
	ARI	Area Name and No. MAMENT RIGHT EFAB AND MLG – 4		Aircraft Serial No.	Date	Total Hrs. This Area	l
Inspect Phase Nos.		Inspection Requirements	Status	Faults and/or Remarks		Action Taken	Initial
ALL	1.	loose mounting and connections.Remove SEU					
ALL	SEU to Electronic Equipment Test Facility (EETF) for testing.						
ALL	3.	DEU and connectors for corrosion, damage, loose mounting and connections. Remove DEU.					
ALL	4.	DEU to Electronic Equipment Test Facility (EETF) for testing.					
ALL	5.	Inspect air hole in air flow adapters on rear of PEU for build-up of dust. Inspect PEU for loose, missing or stripped mounting hardware. Remove PEU.					

РНА	SEI	NO		a Name : IGHT E	and No. FAB AND MLG – 4		Aircraft Serial No.	Dat	е
Inspect Phase Nos.		Inspection Red	quirements	Status	Faults and/or Rema	arks	Action Taken		Initial
ALL	6.	Remove fuseholder caps for F1 through F4. Check for missing, blown, or incorrect fuses.							
ALL	7.	. Connectors J1 through J5 for corrosion and loose connections. Check for loose, bent, burned, or broken pins and sockets.							
ALL	8.	Avioncs equipment for mounting security, external damage, and loose connectors. Wiring harnesses and clamps for chafing, security and deterioration.							
ALL	9.	Wire harnesses for c loose, bent, broken p							
ALL	10.	 Gun control box (GCB) and connectors for corrosion, damage, loose mounting and connectors. 							
ALL	11.	. Turret control box (TCB) and connectors for corrosion, damage, loose mounting, and connections.							

РНА	SE NO	Area Name and No. ARMAMENT RIGHT EFAB AND MLG – 4				Date	е	
Inspect Phase Nos.	e Inspection Requirements			Faults and/or Rema	rks	Action Taken		Initial
ALL	12. Sideloader magazine security, proper oper gears, shaft carriers	ation, broken or bent						

РНА	SE	NO P	HASE	D MAINTENANCE CHECKLIST			
		Area Name and No. LEFT WING AND PYLON – 5		Aircraft Serial No.	Date	Total Hrs. This Area	ı
Inspect Phase Nos.		Inspection Requirements	Status	Faults and/or Remarks		Action Taken	Initial
ALL	1.	Exterior skin for holes, cracks, dents, corrosion, loose or working rivets, and loose or missing hardware.					
ALL	2.	Access panels, fairings, and wing tip door for deformation, cracks, corrosion, loose or working rivets, and loose or missing hardware. Door hinge and latch for damage, binding, security, and proper operation. Seals for wear and deterioration.					
ALL C	3.	Inspect wing forward upper mounting flange area and mounting bolts for security, cracks and corrosion.					
ALL C	4.	. Interior components for mounting security and loose or missing hardware.					
ALL C	5.	Interior structure for cracks, corrosion, loose or working rivets, and loose or missing hardware.					

РНА	SE	NO		a Name a NG AND	and No. PYLONS – 5		Aircraft Serial No.	Date	е
Inspect Phase Nos.		Inspection Re	quirements	Status	Faults and/or Rema	rks	Action Taken		Initial
ALL C	6.	Wing mount fittings for and distortion.	or cracks, corrosion						
ALL C	7.	 Left outboard pylon wiring harnesses for security, chafing, and loose connections. Hydraulic and fuel lines for security, chafing, and leakage. 							
ALL	8.	 Left inboard pylon wiring harnesses for security, chafing, and loose connections. Hydraulic and fuel lines for security, chafing, and leakage. 							
ALL	9.	Wing wiring harnesse and proper connection	es for security, chafing, on.						
2,4 C	Spars for cracks, corrosion, and loose or working rivets. Hydraulic lines for leakage, chafing, and clamping security. Wiring harnesses for chafing and clamping security.								
2,4 C	Pitot/static lines for cracks, chafing, and mounting security.								

PHA	SE	NO	HASED I	MAINTENANCE CHECKLIST			
	ARM	Area Name and No. IAMENT LEFT WING AND PYLON – 5		Aircraft Serial No.	Date	Total Hrs. This A	rea
Inspect Phase Nos.		Inspection Requirements	Status	Faults and/or Remarks	A	ction Taken	Initial
ALL	1.	Left outboard installed pylons and racks for cracks, dents, distortion, and loose or missing fastners. Pylon and rack mount fittings for cracks and distortion. Mounting bolts for security.					
ALL	Pylon fairings and actuators for evidence of damage and hydraulic fluid leakage.						
ALL	3.	Access P3. Pylon PIU bracket for cracks, corrosion, or warping.					
ALL	4.	Check pylon connectors for corrosion, damage, and loose mounting connections.					
ALL	5.	Check pylon rack inserts, bushings, and bearings for visible damage.					
ALL	6.	Check pylon rack mounting bushings.					
ALL	7.	Check pylon hollow pin.					

[&]quot;FOD REMINDER" Check work area for tools and parts after completion of maintenance and inspections.

2-45 C1

PHA	SEI	NO		a Name a FT WING	and No. G AND PYLONS – 5		Aircraft Serial No. Date		е
Inspect Phase Nos.		Inspection Re	quirements	Status	Faults and/or Rema	rks	Action Taken		Initial
ALL	8.	Check pylon frame assembly for cracks on both inboard and outboard sides. Use magnifier around pivot pin area. No cracks allowed.							
ALL C	9.	Pylon ejector assembly for corroded piston and cartridge holder.							
ALL	10.	. Check pylon rack articulation.							
ALL	11.	1. Check pylon rack for allowable release torque. Position rack hooks to the indicated hook-locked position. Move rack hooks to the indicated hook-unlocked position. Note torque reading at release point. Use dial indicating torque wrench. Rack is operational if torque reading is equal to or less than 120 inch-pounds. Repair rack if reading to open rack hooks is greater than 120 inch-pounds.							
ALL C	12.	 Left inboard installed pylons and racks for cracks, dents, distortion, and loose or missing fastners. Pylon and rack mount fittings for cracks and distortion. Mounting bolts for security. 							

[&]quot;FOD REMINDER" Check work area for tools and parts after completion of maintenance and inspections. 2-46 C1

РНА	SE	NO		a Name FT WIN	and No. G AND PYLONS – 5		Aircraft Serial No.	Date	е
Inspect Phase Nos.		Inspection Red	quirements	Status	Faults and/or Rema	rks	Action Taken		Initial
ALL	13.		ctuators for evidence of						
		damage and hydraul	ic fluid leakage.						
ALL	14.								
	corrosion, or warping.								
ALL	15.	 Check pylon connectors for corrosion, damage, and loose mounting connections. 							
ALL	16. Check pylon rack inserts, bushings, and bearings for visible damage.								
ALL	17.	Check pylon rack mo	ounting bushings.						
ALL	18.	Check pylon hollow p	oin.						
ALL	10	Charle miles from a	an amply for an also an						
ALL	19.	both inboard and out	ssembly for cracks on board sides. Use						
			ot pin area. No cracks						
		allowed.							
ALL	20.	Pylon ejector asseml	bly for corroded piston						
		and cartridge holder.							
С		and cartriage noider.							

[&]quot;FOD REMINDER" Check work area for tools and parts after completion of maintenance and inspections. 2-47

РНА	SE	NO		a Name a FT WING	and No. G AND PYLONS – 5		Aircraft Serial No.	Date	е
Inspect Phase Nos.		Inspection Red	quirements	Status	Faults and/or Rema	rks	Action Taken		Initial
ALL	21.	Check pylon rack art	iculation.						
ALL	22. Check pylon rack for allowable release torque. Position rack hooks to the indicated hook-locked position. Move rack hooks to the indicated hook-unlocked position. Note torque reading at release point. Use dial indicating torque wrench. Rack is operational if torque reading is equal to or less than 120 inch-pounds. Repair rack if reading to open rack hooks is greater than 120 inch-pounds.								
ALL	23.	Perform missile laune	cher inspection.						
ALL	23a.	. Inspect rocket launcl	her.						
ALL	24.	Rocket pod for wear, bends, and dents in t	cracks, deterioration, tubes.						
ALL	25.								
ALL	26.	Igniter arms for prope	er operation.						
ALL	27. Lubricate rocket pods.								

[&]quot;FOD REMINDER" Check work area for tools and parts after completion of maintenance and inspections.

2-48 C1

РНА	SE	NO P	HASEI	D MAINTENANCE CHECKLIST			
		Area Name and No. RIGHT WING AND PYLONS – 6	Aircraft Serial No.		Date	Total Hrs. This Area	ı
Inspect Phase Nos.		Inspection Requirements	Status	Faults and/or Remarks		Action Taken	Initial
ALL	1.	Exterior skin for holes, cracks, dents, corrosion, loose or working rivets, and loose or missing hardware.					
ALL	2.	Access panels, fairings, and wing tip door for deformation, cracks, corrosion, loose or working rivets, and loose or missing hardware. Door hinge and latch for damage, binding, security, and proper operation. Seals for wear or deterioration.					
ALL C	3.	Inspect wing forward upper mounting flange area and mounting bolts for security, cracks and corrosion.					
ALL C	4.	Interior components for mounting security and loose or missing hardware.					
ALL C	5.	 Interior structure for cracks, corrosion, loose or working rivets, and loose or missing hardware. 					

"FOD REMINDER" Check work area for tools and parts after completion of maintenance and inspections.

PHA	SE	NO		a Name NG AND	and No.) PYLONS – 6		Aircraft Serial No.	Dat	е
Inspect Phase Nos.		Inspection Requirements		Status	Faults and/or Rema	arks	Action Taken		Initial
ALL C	6.	Wing mount fittings for cracks, distortion and corrosion.							
ALL	7.	Right outboard pylon wiring harnesses for security, chafing, and loose connections. Hydraulic and fuel lines for security, chafing, and leakage.							
ALL	8.	Right inboard pylon wiring harnesses for security, chafing, and loose connections. Hydraulic and fuel lines for security, chafing, and leakage.							
ALL	9.	Wing wiring harnesses for security, chafing, and proper connection.							
2,4 C	10.	 Spars for cracks, corrosion, and loose or working rivets. Hydraulic lines for leakage, chafing, and clamping security. Wiring harnesses for chafing and clamping security. 							

"FOD REMINDER" Check work area for tools and parts after completion of maintenance and inspections.

PHA	SE NO	Area Name and No. RIGHT WING AND PYLONS – 6				Date	е	
Inspect Phase Nos.	Inspection Requirements		Status	Faults and/or Rema	arks Action Taken			Initial
2,4		3,						
С	mounting security.							

"FOD REMINDER" Check work area for tools and parts after completion of maintenance and inspections.

РНА	SE	NO P	HASE	D MAINTENANCE CHECKLIST			
А	RMA	Area Name and No. AMENT RIGHT WING AND PYLONS – 6		Aircraft Serial No.	Date	Total Hrs. This Area	l
Inspect Phase Nos.		Inspection Requirements	Status	Faults and/or Remarks	Action Taken		Initial
	,	District description of the second sector					
ALL	1.	Right outboard installed pylons and racks for cracks, dents, distortion, and loose or					
		missing fastners. Pylons and rack mount					
		fittings for cracks and distortion. Mounting					
		bolts for security.					
ALL	2.	Pylon fairings and actuators for evidence of					
		damage and hydraulic fluid leakage.		1			
ALL	3.	Access P3. Pylon PIU bracket for cracks,					
		corrosion, and warping.					
		· -					
ALL	4.	Check pylon connectors for corrosion,					
		damage, and loose mounting connections.					
ALL	5.	Check pylon rack inserts, bushings, and					
,	0.	bearings for visible damage.					
		· ·					
ALL	6.	Check pylon rack mounting bushings.					
A	7	Chack pulsa hallow nia					
ALL	7.	. Check pylon hollow pin.					

"FOD REMINDER" Check work area for tools and parts after completion of maintenance and inspections. 2-53 C1

PHA	SE	NO		a Name a SHT WIN	and No. IG AND PYLONS – 6		Aircraft Serial No.	Dat	е
Inspect Phase Nos.		Inspection Re	quirements	Status	Faults and/or Remai	rks	Action Taken		Initial
ALL	8.	Check pylon frame assembly for cracks on both inboard and outboard sides. Use magnifier around pivot pin area.							
ALL C	9.	Pylon ejector assembly for corroded piston and cartridge holder.							
ALL	10.	Check pylon rack articulation.							
ALL	11.	1. Check pylon rack for allowable release torque. Position rack hooks to the indicated hook-locked position. Move rack hooks to the indicated hook-unlocked position. Note torque reading at release point. Use dial indicating torque wrench. Rack is operational if torque reading is equal to or less than 120 inch-pounds. Repair rack if reading to open rack hooks is greater than 120 inch-pounds.							
ALL C	12.	 Right inboard installed pylons and racks for cracks, dents, distortion, and loose or missing fastners. Pylon and rack mount fittings for cracks and distortion. Mounting bolts for security. 							

[&]quot;FOD REMINDER" Check work area for tools and parts after completion of maintenance and inspections.

2-54 C1

РНА	SE	NO		a Name a HT WIN	and No. IG AND PYLONS – 6		Aircraft Serial No.	Date	е
Inspect Phase Nos.		Inspection Red	quirements	Status	Faults and/or Remai	rks	Action Taken		Initial
ALL	13.	Pylon fairings and ac damage and hydraul	ctuators for evidence of ic fluid leakage.						
ALL	14.	 Access P3. Pylon PIU bracket for cracks, corrosion, or warping. 							
ALL	15.	 Check pylon connectors for corrosion, damage, and loose mounting connections. 							
ALL	16.	 Check pylon rack inserts, bushings, and bearings for visible damage. 							
ALL	17.	Check pylon rack mo	ounting bushings.						
ALL	18.	Check pylon hollow p	pin.						
ALL	Check pylon frame assembly for cracks on both inboard and outboard sides. Use magnifier around pivot pin area.								
ALL C	Pylon ejector assembly for corroded piston and cartridge holder.								

РНА	SE	NO		a Name a BHT WIN	and No. IG AND PYLONS – 6		Aircraft Serial No.	Date	е
Inspect Phase Nos.		Inspection Red	quirements	Status	Faults and/or Remar	ks	Action Taken		Initial
ALL	21.	Check pylon rack arti	iculation.						
ALL	22. Check pylon rack for allowable release torque. Position rack hooks to the indicated hook-locked position. Move rack hooks to the indicated hook-unlocked position. Note torque reading at release point. Use dial indicating torque wrench. Rack is operational if torque reading is equal to or less than 120 inch-pounds. Repair rack if reading to open rack hooks is greater than 120 inch-pounds.								
ALL	23.	Perform missile laund	cher inspection.						
ALL	23a.	. Inspect rocket launch	her.						
ALL	24.	Rocket pod for wear, bends, and dents in t	cracks, deterioration, tubes.						
ALL	25. Rocket pod connectors for wear or broken pins.								
ALL	26. Igniter arms for proper operation.								
ALL	27. Lubricate rocket pods.								

[&]quot;FOD REMINDER" Check work area for tools and parts after completion of maintenance and inspections.

2-56 C1

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PHA	ASE NO	PHASED MA	AINTENANCE CHECKLIST			
	Area Name and No. LEFT ENGINE AND NOSE GEARBOX	<u> </u>	Aircraft Serial No.	Date	Total Hrs. Th	is Area
Inspect Phase Nos.		Status	Faults and/or Remarks	A	ction Taken	Initial
ALL	Exterior skin for holes, cracks, do corrosion, loose or working rivets or missing hardware.					
ALL	2. Access panels, fairings, and doc deformation, cracks, corrosion, le working rivets, and loose or miss hardware. Door hinges, struts, a for damage, binding, and securit for security and proper operation wear and deterioration.	oose or sing nd supports y. latches				
ALL C	Navigation, and anti-collision ligh corrosion, loose or missing hard security. lenses for cracks, loose discoloration. Wiring harness for and clamping security.	ware, and ness, and				
ALL	IR suppressor nozzle mating sur worn, and damaged orange seal					
ALL	Interior structure for cracks, correct or working rivets, and loose or management hardware.					

РНА	SE	NO		a Name a AND NO	and No. OSE GEARBOX – 7		Aircraft Serial No.	Dat	е
Inspect Phase Nos.		Inspection Re	quirements	Status	Faults and/or Remar	·ks	Action Taken		Initial
2,4 C	6.	Particle separator du and deformation.	act for cracks, dents,						
ALL	7.	 IR suppressor nozzles and radiation shields for cracks, dents, deformation, and security. Supports for damage and security. 							
2,4	8.	B. Main transmission heat exchangers and engine cooloing louvers for cracks, deformation, delamination, distortion, broken and loose or working rivets. Louver plates for looseness and lost motion.							
ALL	9.	 Engine air inlet for cracks, distortion, security, corrosion, deteriorated, torn and split seals. Loose or working rivets and screws. Interior for cleanliness. 							
2,4	Drain and service engine starter. Check for cracks and loose or missing hardware.								
2,4	11.	Engine wiring harnesses for loose connections, chafing, and deterioration.							

PHA	SE	NO		a Name a AND NO	and No. OSE GEARBOX – 7		Aircraft Serial No.	Date	е
Inspect Phase Nos.		Inspection Red	quirements	Status	Faults and/or Remai	rks	Action Taken		Initial
ALL C	12.		I cables and brackets , mounting security, and dware.						
ALL C	13. Primary exhaust nozzle for cracks.								
2,4 C	14. Visually inspect engine mounts for cracks, deformation, loose bushings, corrosion, security, and loose or missing hardware. Pins and expanding bolts for wear and looseness.								
ALL	15.	Nose gearbox wiring chafing, and deterior	for loose connections, ation.						
ALL C	Remove and clean nose gearbox chip detector. Check for insulation damage.								
ALL C	17.	Remove and clean n	ose gearbox breather.						
ALL C	18. Change nose gearbox lube , oil, and filter.								

"FOD REMINDER" Check work area for tools and parts after completion of maintenance and inspections.

2-59 C1

РНА	SEI	NO		a Name a AND NO	and No. DSE GEARBOX – 7		Aircraft Serial No.	Date	е
Inspect Phase Nos.		Inspection Red	quirements	Status	Faults and/or Rema	ırks	Action Taken		Initial
ALL C	 Nose gearbox oil pressure switch, pressure transducer, and temperature probe for insulation damage, oil leakage, and security. Harness splices for security. 		perature probe for bil leakage, and security.						
ALL C	20.	20. Nose gearbox lube oil level sight gage for cleanliness, leakage, and security.							
ALL C	21.	 Nose gearbox and lube oil pump housings for cracks, distortion, leakage, and security. 							
2,4 C	22.	Nose gearbox mount is 60 inch/lbs.	ting bolts, verify torque						
2,4 C	23.		g fins and shroud for corrosion, and security. osion and damage.						
ALL	24. Nose gearbox drive shaft and couplings for nicks, dents, scratches, and security.								

РНА	SE	NO P	HASEI	D MAINTENANCE CHECKLIST			
ı	RIGH	Area Name and No. IT ENGINE AND NOSE GEARBOX – 8		Aircraft Serial No.	Date	Total Hrs. This Area	
Inspect Phase Nos.		Inspection Requirements	Status	Faults and/or Remarks		Action Taken	
ALL	1.	Exterior skin for holes, cracks, dents, corrosion, loose or working rivets, and loose or missing hardware.					
ALL	2.	Access panels, fairings, and doors for deformation, cracks, corrosion, loose or working rivets, and loose or missing hardware. Door hinges, struts, and supports for damage, binding, and security. Latches for security and proper operation. Seals for wear or deterioration.					
ALL	3.	Navigation and anti-collision lights for corrosion, loose or missing hardware, and security. Lenses for cracks, looseness, and discoloration. Wiring harness for chafing and clamping security.					
ALL	4.	IR suppressor nozzle mating surface for worn and damaged orange seals.					

РНА	SE	NO		ea Name and No. E AND NOSE GEARBOX – 8		Aircraft Serial No.	Date	е	
Inspect Phase Nos.		Inspection Re	quirements	Status	Faults and/or Rema	rks	Action Taken		Initial
ALL	5.	Interior etructure for	cracks, corrosion. loose						
ALL	5.	or working rivets, and							
		hardware.	g						
0.4		D (1)							
2,4	6.	Particle separator du and deformation.	ict for cracks, dents,						
С		and deformation.							
ALL	7.	IR suppressor nozzles and radiation shields							
			formation, and security.						
		Supports for damage	e and security.						
2,4	8.	Main transmission he	eat exchanger and						
,		engine cooling louve	rs for cracks,						
		deformation, delamin							
		broken or loose rivets looseness and lost m							
		iooseness and lost if	IOUOH.						
ALL	9.	Engine air inlet for cr	acks, distortion,						
		security, corrosion, d	leteriorated, torn and						
		split seals. Loose or							
		screws. Interior for cl	leanliness.						

РНА	SE	NO		ea Name and No. E AND NOSE GEARBOX – 8			Aircraft Serial No.	Date	е
Inspect Phase Nos.		Inspection Red	quirements	Status	Faults and/or Rema	rks	Action Taken		Initial
2,4	10.	Drain and service en cracks and loose or r	gine starter. Check for missing hardware.						
2,4	Engine wiring harnesses for loose connections, chafing, and deterioration.								
2,4 C	12.	 Engine power control cables and brackets for cracks, corrosion, mounting, security, and loose or missing hardware. 							
ALL C	13.	Primary exhaust noz	zle for cracks.						
2,4 C	14.	Visually inspect engited deformation, loose be security, and loose of Pins and expanding looseness.	r missing hardware.						
ALL	Nose gearbox wiring for loose connections, chafing, and deterioration.								

РНА	SE	NO		a Name a AND N	and No. OSE GEARBOX – 8		Aircraft Serial No.	Dat	е
Inspect Phase Nos.		Inspection Red	quirements	Status	Faults and/or Rema	rks	Action Taken		Initial
ALL C	16.	Remove and clean n detector. Check for in							
ALL C	17.	17. Remove and clean nose gearbox breather.							
ALL C	18.	Change nose gearbo	ox lube, oil and filter.						
ALL C	19.	 Nose gearbox oil pressure switch, pressure transducer, and temperature probe for insulation damage, oil leakage, and security. Harness splices for security. 							
ALL C	20.	Nose gearbox lube o cleanliness, leakage,							
ALL C	21.	Nose gearbox and lu for cracks, distortion,	be oil pump housing leakage, and security.						
2,4 C	22.	Nose gearbox mount is 60 inch/lbs.	ting bolts, verify torque						

PHA	DUAGENIO		a Name and No. AND NOSE GEARBOX – 8			Aircraft Serial No.	Date	е	
Inspect Phase Nos.		Inspection Requirements		Status	Faults and/or Rema	ırks	Action Taken		Initial
2,4 C	23.	Nose gearbox cooling fins and shroud for cracks, deformation, corrosion, and security. Fan impeller for corrosion and damage.							
ALL	24. Nose gearbox drive shaft and couplings for nicks, dents, scratches, and security.								

PHA	SE	NO P	HASED	MAINTENANCE CHECKLIST			
		Area Name and No. MAIN TRANSMISSION - 9		Aircraft Serial No.	Date	Total Hrs. This Area	a
Inspect Phase Nos.		Inspection Requirements	Status	Faults and/or Remarks	A	Action Taken	Initial
ALL	1.	Exterior skin for holes, cracks, dents, corrosion, loose or working rivets, and loose or missing hardware.					
ALL	2.	Access panels, fairings, and doors for deformation, cracks, corrosion, loose or working rivets, and loose or missing hardware. Door hinges for damage, binding, and security. Latches for security and proper operation. Seals and copper coat for wear or deterioration.					
ALL	3.	Interior components for mounting security and loose or missing hardware.					
ALL	4.	Interior structure for cracks, corrosion, loose or working rivets, and loose or missing hardware.					
ALL	5.	Transmission housing and cover for cracks, oil leakage, and evidence of overheating (discoloration).					

PHA	SE	NO		a Name a RANSM	and No. ISSION – 9		Aircraft Serial No. Date		
Inspect Phase Nos.		Inspection Red	quirements	Status	Faults and/or Rema	rks	Action Taken		Initial
ALL	6.	Torque check upper case nuts to 190 in. lbs.							
ALL C	7.	 Remove and clean chip detectors. Check for insulation and damage. 							
2,4	8.	Wiring harnesses for chafing, or deteriorat							
ALL C	9.	Replace accessory p	oump oil filter. Remove reen.						
ALL C	10.	Clean transmission b	preathers.						
ALL C	11.	Change transmission lube oil and filters.							
2,4	12.	Oil pressure switches transducers, tempera magnetic pickup for i leakage, and security security.	ature probes, and nsulation damage,						

РНА	SE	NO		a Name RANSM	and No. IISSION – 9		Aircraft Serial No.	Dat	е
Inspect Phase Nos.		Inspection Red	quirements	Status	Faults and/or Rema	ırks	Action Taken		Initial
2,4	13.	13. Reaction bar for cracks and corrosion.							
ALL	14.	Reaction bar for signs of wear, presence of metal fragments, elongated hole, and torque striping on two clamping bolts.							
ALL C	15.	. Lube oil level sight gages for cleanliness, leakage, and security.							
2,4 C	16.	Input shaft and coupl installation. Verify tor							
ALL C	17.		anifold for leakage, nections, and security. llet check valve filter for						
ALL C	18.	Generators for dama and cracked or broke	ged insulation, security, en housings.						

РНА	SE	NO		a Name RANSM	and No. IISSION – 9		Aircraft Serial No.	Date	е
Inspect Phase Nos.		Inspection Red	quirements	Status	Faults and/or Rema	rks	Action Taken		Initial
2,4	19.	Transformer/rectifiers overheating (discolor							
ALL C	20.	Anti-collision light power supply for corrosion and loose or missing hardware.							
ALL C	21.		n. Upper and lower rod eized bearings. Rubber and deterioration. er control linkage						
ALL C	22.	Main rotor mast supp bending, distortion, a Transmission deck for looseness at lower e	and security. or distortion and						
ALL C	23.	the upper portion of t	rt mount and upper s for security. Inspect the mast base support he four lighting holes,						

РНА	\SE I	NO		a Name a RANSM	and No. ISSION – 9	Aircraft Serial No.	Dat	е	
Inspect Phase Nos.		Inspection Red	quirements	Status	Faults and/or Rema	arks	Action Taken		Initial
2,4 C	24.	Static mast attaching hardware for proper torque (use PLI washer method).							
ALL C	25.	Flight control rod and rod ends for dents, cracks, corrosion, security, and worn or seized bearings and bushings.							
ALL	26.	Flight control bellcrar corrosion, and securi mounting security. Pi looseness.	ity. Brackets for						
2,4 C	27.	cables, supports, clar cracks and bends. Roworn and seized bea	rings. Bellcranks for worn bushings, loose						

PHA	SE	NO P	HASED N	MAINTENANCE CHECKLIST			
		Area Name and No. MAIN ROTOR – 10	Aircraft Serial No.		Date	Total Hrs. This Area	1
Inspect Phase Nos.		Inspection Requirements	Status	Faults and/or Remarks	A	Action Taken	Initial
ALL	1.	Blade spars and root finger doublers for delamination.					
ALL C	2.	Blade leading edge for cracks, dents, distortion, and erosion. Leading edge tips for loose or missing hardware. Trailing edge for cracks and voids.					
ALL	3.	Perform coin tap test on main rotor blades.					
ALL	4.	Blade root bushings for cracks, distortion, and security.					
ALL C	and corrosion. Mounting bolts for security.						

РНА	SEI	NO		a Name a N ROTO			Aircraft Serial No.	Date	Э
Inspect Phase Nos.		Inspection Red	quirements	Status	Faults and/or Rema	rks	Action Taken		Initial
2,4 C	6.	Main rotor hub for cracks, distortion, corrosion, and security. Hub lower grease seal for leakage.							
2,4 C	7.	Feathering bearing housings and retaining nuts for cracks, corrosion, and security. Feathering bearings for radial looseness.							
2,4 C	8.	Blade droop stop striker plates for distortion and cracks. Droop stop plungers, return springs, and stop ring for cracks and deformation. Plungers, rollers, and stop ring for wear and play.							
2,4 C	9.	Pitch housings for cra corrosion, and securi for pitch link rod end	ty. Pitch housing ears						
2,4 C	10.	 Lead-lag links for cracks, distortion, and corrosion. Link-to-damper rod ends for bearing damage. Rod ends for worn and seized bearings. 							
2,4 C	11.	Lead-lag dampers for security and loose, debonded, or deteriorated elastic material.							

РНА	SE	NO			a Name and No. IN ROTOR – 10		Aircraft Serial No.	Dat	е
Inspect Phase Nos.		Inspection Red	quirements	Status	Faults and/or Rema	arks	Action Taken		Initial
ALL C	12.	Blade strap packs for cracks, buckling, breakage or horizontal displacement. (Check both ends of each strap).							
2,4 C	13.	Static discharge brushes for fraying, distortion, cracked and broken holders or springs. Brushes for even contact with mast.							
ALL C	14.	corrosion. Rod ends	, distortions, and for bearing damage or wer rod end clamp-up						
ALL C	15.	Swashplate uniball for flaked or worn-throug grease leakage.	or cracks, grooving, gh plating. Bearing for						
ALL C	16.	5. Rotating swashplate for cracks, corrosion, grease leakage, and security of lower seal. Pitch link connection bosses for bending, misalignment, and worn or loose bushings.							

РНА	SE NO		a Name a N ROTO	and No. DR – 10		Aircraft Serial No.	Dat	е
Inspect Phase Nos.	Inspection Re	quirements	Status	Faults and/or Rema	arks	Action Taken		Initial
ALL C	corrosion. Lateral an	 Stationary swashplate for cracks and corrosion. Lateral and torque connection bosses for bending, misalignment, and worn or loose bushings. 						
ALL C	18. Rotating scissors for security. Pivot bearin	cracks, corrosion, and gs for wear.						
ALL C	19. Longitudinal and late cracks, dents, scratc Attachment bolts for looseness.							
ALL C	20. Longitudinal, lateral, bellcracks for cracks corrosion. Check floato rod ends.							
2,4 C	21. Mixer supports for cr corrosion. Mixer atta cracks, corrosion, an worn or seized bearing	chment bolts for and security. Check for						

РНА	SEI	NO		a Name a N ROTO			Aircraft Serial No.	Date	е
Inspect Phase Nos.		Inspection Red	quirements	Status	Faults and/or Rema	arks	Action Taken		Initial
ALL C	22.	Lower shoes for worn scissor bearings and worn plunger bearings.							
ALL C	23.	Main rotor hub nut for corrosion, stripped threads, cracks, or failure. Magnetic particle inspect hub nut.							
ALL C	24.	Check rotor hub stati	c droop angle.						
2,4 C	25.		nplate sliding surface						
ALL C	26.	Derotation unit for da deformation (If install	mage, looseness, and ed).						
ALL C	27.	Mast base flange for cracks, corrosion, and security.							

PHA	DUIA OF NO			a Name and No. IN ROTOR – 10			Aircraft Serial No.	Date	е
Inspect Phase Nos.	ase Inspection Requirements			Status	Faults and/or Rema	arks	Action Taken		Initial
2,4 C	28.	28. Check main rotor upper bearing using 12X magnifying glass. Inspect bearing for burnt grease and contamination. Repack bearing.							
ALL C	29. Main rotor gear shaft for cracks, distortion, corrosion, and for chipped, broken, scored or worn spline teeth.								

РНА	SE NO P	HASE	MAINTENANCE CHECKLIST			
	Area Name and No. ARMAMENT MAIN ROTOR – 10		Aircraft Serial No.	Date	Total Hrs. This Area	1
Inspect Phase Nos.	Inspection Requirements	Status	Faults and/or Remarks	Action Taken		Initial
C C	1. Remove the 12 MMA to DU attachment bolts and separate the MMA from the DU. Visually inspect at 10X magnification minimum, the DU attachment bolt holes (12) and nut plate cage rivet holes (24) for cracks on the bottom surface of the stationary pedestal. Also visually inspect the inside surface of the stationary pedestal for cracks (opposite the lower azimuth bearing journals).					
ALL C	2. Verify integrity of torque stripes on baseplate attachment to pedestal shelf (12 screws external). Discrepant torque stripe: None allowed. If found correct. Torque to 115 inch-pounds. Use torque wrench. Apply torque stripe.					
ALL C	Unfasten the 26 Radome/Aft Dome bolts and remove the Radome.					
ALL C	Unfasten the 25 Aft Dome/Baseplate bolts and open the Aft Dome for internal inspection.					

РНА	SE	NO		a Name NT MAIN	and No. N ROTOR – 10		Aircraft Serial No.	Date	е
Inspect Phase Nos.		Inspection Red	quirements	Status	Faults and/or Rema	rks	Action Taken		Initial
ALL C	5.	Twenty-two Hub Coll Twenty-four Hub Coll Any fastner with torque stripe chafed, misaling should be inspected. Visually inspect all ur	tached wave guides by the Line (LRM) from the o inspect and verify the following: lar/Rotary Tube Bolts; llar/Baseplate Bolts. ue stripe damage (i.e. gned, or missing) and fully torqued. hoblted and unfastened and around bolt/fastner cion. ube Bolts discrepant allowed. If found inch-pounds. Use y torque stripe. e Bolts discrepant allowed. If found bolts to 115 #10 bolts to 50						
				l .	I		I .		ı

РНА	SE	NO		a Name a NT MAIN	and No. I ROTOR – 10		Aircraft Serial No.	Date	е
Inspect Phase Nos.		Inspection Red	quirements	Status	Faults and/or Rema	rks	Action Taken		Initial
Nos.	6.	Visually verify torque following external bot torque stripe. None a re—torque and apply Four Azimuth Drive Coplaces. (Torque to 25 Six Azimuth Drive Electronic Screws. (Torque to 25 Four RFI Antenna Cato 79 inch-pounds) Four RFI Receiver Companies to 79 inch-pounds) Four RFI Receiver But (On units where instanch-pounds). Two Time Meter Screeninch-pounds) Ten Inertial Particle Summer (Torque to 50 inch-pounds)	stripe integrity for the lt/screws: Discrepant allowed. If found, torque stripe. Captive Screws – two inch-pounds) ectronic Unit Captive 5 inch-pounds) aptive Screws. (Torque to 50) Ews. (Torque to 50)	Status	Faults and/or Remains	rks	Action Taken		Initial
		Four Cooling Fan Screws. (Torque to 50 inch-pounds)							

РНА	SE	NO		a Name a NT MAIN	and No. I ROTOR – 10		Aircraft Serial No.	Dat	е
Inspect Phase Nos.		Inspection Red	quirements	Status	Faults and/or Rema	ırks	Action Taken		Initial
ALL C	7.	Radome Shell: Check for evidence of fatigue damage appearing as delamination, cracks, or grazing at the bolting flange; area of the top flat surface from the captive stud bolts, eight inches forward; and area eight inches wide along the tangent line between the top flat surface and the upper corner radius, four inches either side of the tangent line. Any noticable flaw sized less than 1/2 inch diameter should be reported to AMCOM Engineering and subsequently monitored. A flaw sized greater than 1/2 inch requires repair by the MMA manufacturer and reported to USAAMCOM							
		Engineering.							
ALL C	8.	Install Transmitter.							
ALL	9.	Close AFT dome.							
С									
ALL	10.	Install Radome.							
С									

РНА	SEI	NO	1	a Name a NT MAIN	and No. I ROTOR – 10		Aircraft Serial No.	Dat	е
Inspect Phase Nos.		Inspection Red	quirements	Status	Faults and/or Rema	ırks	Action Taken		Initial
C C	11.	I. On 12 MMA to DU attachment bolts, check for minimum run in torque of 6.5 in–lbs. Torque shall be applied in stages to pairs of bolts diametrically opposed at 90 degrees (at right angle) to the first pair. This shall be followed by pairs of bolts between first two pairs. The first stage shall snug the head of the bolt against the contacting structure. The second and third stages shall torque the bolts to 50 percent and 100 percent of specified installation torque consecutively.							
ALL C	12.	Apply torque stripes a following bolts and so 12 MMA to DU attack 12 screws on basepla pedestal shelf.	crews: hment bolts;						

РНА	SE NO	Area Name and No. ARMAMENT MAIN ROTOR – 10				Date		
Inspect Phase Nos.	Inspection Requirements		Status	Faults and/or Rema	rks	Action Taken		Initial
	13. Deleted.							

РНА	SE	NO P	HASE	D MAINTENANCE CHECKLIST			
		Area Name and No. AFT EQUIPMENT BAY – 11		Aircraft Serial No.	Date	Total Hrs. This Area	
Inspect Phase Nos.		Inspection Requirements	Status	Faults and/or Remarks		Action Taken	Initial
ALL	1.	Exterior skin for holes, cracks, dents, corrosion, loose or working rivets, and loose or missing hardware. Handholds for damage, structural integrity, and mounting security.					
ALL	2.	Access doors for deformation, cracks, corrosion, loose or working rivets, and loose or missing hardware. Door hinges for damage, binding, and security. Latches and lanyards for security and proper operation. Seals for wear and deterioration.					
ALL C	3.	Structural mating surfaces and fairings for worn or deteriorated seals and copper coat.					
ALL C	4.	Tail rotor drive shaft and coupling bolts for proper installation. Verify torque on bolts.					

PHA	SE	NO		a Name and No. UIPMENT BAY – 11			Aircraft Serial No.	Date	е
Inspect Phase Nos.		Inspection Red	quirements	Status	Faults and/or Rema	rks	Action Taken		Initial
ALL C	5.	Tail rotor drive shaft forward and aft hanger bearings for cracks, corrosion, mounting security, and for radial looseness and smooth operation. Hanger bearings for smooth rotation. Verify drive shaft forward and aft hanger bearing supports for torque of attaching bolts.							
2,4 C	6.	Main transmission heat exchangers and louver actuators for cracks, deformation and security.							
2,4 C	7.	Fire extinguisher for a security. Cartridges, for cracks, distortion and extinguisher charge produced discharge—indicating	fittings and valves for I security. Check fire pressure. Inspect						
2,4 C	8.	Fire extinguisher outl for cracks, dents, nic distortion, and securi tubes around "B" nut	ty. Inspect interior of						
2,4 C	9.	Fire extinguisher systematic interior corrosion, pitt evaporation							

"FOD REMINDER" Check work area for tools and parts after completion of maintenance and inspections.

2-86 C1

PHA	SE	NO		Area Name and No. AFT EQUIPMENT BAY - 11			Aircraft Serial No.		е
Inspect Phase Nos.		Inspection Red	quirements	Status	Faults and/or Rema	rks	Action Taken		Initial
ALL	10.	cracks. delamination, seal damage, loose or missing fastners, and security. Check for APU closure, chafing between bulkhead and deck.							
ALL	11.		coupling for cracks, rosion, and evidence of						
ALL	12.	APU drive shaft coup torque.	oling bolts for proper						
ALL	13.	APU mounts for crace and corrosion. Remo the APU hardware, n lugs/surfaces for crace distortion or elongation	ove APU and inspect nounts, mounting cks, corrosion, and						
ALL	14.	APU starter for crack security.	s, leakage and						
ALL	15.	APU combustor for c burned-through area							

PHA	SE	PHASE NO AFT EQ		a Name and No. UIPMENT BAY – 11			Aircraft Serial No.	Date	
Inspect Phase Nos.		Inspection Rec	quirements	Status	Faults and/or Rema	rks	Action Taken		Initial
2,4	16.	APU exhaust fairing tand loose or missing							
ALL	17. Inspect APU clutch friction disc for wear.								
2,4 C	18. Change APU oil filter.								
ALL	19.	Change APU fuel filte	er.						
ALL C	20.	or seized bearings. P	ods. Rod ends for worn Push-pull support nd security. Bellcranks worn or seized						
ALL C	21.	Left and right engine cracks, distortion, cormissing hardware, ar	rrosion, loose or						

"FOD REMINDER" Check work area for tools and parts after completion of maintenance and inspections.

2-88 C1

PHA	PHASE NO AFT EC			ea Name and No. PUIPMENT BAY – 11			Aircraft Serial No.	Dat	е
Inspect Phase Nos.		Inspection Re	quirements	Status	Faults and/or Rema	rks	Action Taken		Initial
ALL C	22.	 Catwalk and supporting structure for cracks, debonding, distortion, loose or missing hardware, and security. Walk-way coating for peeling and deterioration. 							
ALL	23.	Utliity accumulator tubing for leakage and security.							
ALL C	24.	Utility hydraulic mani corrosion, loose con Sight gage for prope manifold air inlet che cleanliness.	nections, and security. r fluid level. Check						
ALL	25.	Hydraulic manifold fil popped buttons.	Iter indicators for						
ALL	26.	Primary and utility hy panels for security. F couplings for fluid lea	vdraulic ground service Panel quick-disconnect akage.						
ALL	27.	Replace primary GS	E panel fluid filter.						

PHA	DULA OF NO		ea Name and No. PUIPMENT BAY – 11			Aircraft Serial No.		е	
Inspect Phase Nos.		Inspection Requirements		Status	Faults and/or Rema	rks Action Taken			Initial
ALL	28.	Vent utility accumulator hydraulic pressure by activating emergency hydraulic system and operating flight controls.							
ALL	29.	9. Verify utility accumulator nitrogen pressure.							

PHA	SE	NO P	HASED	MAINTENANCE CHECKLIST			
	N	Area Name and No. MID AND LOWER FUSELAGE – 12	Aircraft Serial No.		Date	Total Hrs. This Area	l _
Inspect Phase Nos.		Inspection Requirements	Status	Faults and/or Remarks	Action Taken		Initial
ALL	1.	Exterior skin for holes, cracks, dents, corrosion, loose or working rivets, and loose or missing hardware. Steps for damage, structural integrity, and mounting security.					
ALL	2.	Access panels, fairings, and doors for deformation, cracks, corrosion, loose or working rivets, and loose or missing hardware. Door hinges for damage, binding, and security. Latches for security and proper operation. Seals and copper coat for wear and deterioration. Drain holes for obstructions.					
	3.	Deleted.					
ALL	4.	Interior components for mounting security and loose or missing hardware.					
ALL	5.	Interior structure for cracks, corrosion, loose or working rivets, and loose or missing hardware.					

РНА	SEI	NO		ea Name and No. OWER FUSELAGE – 12			Aircraft Serial No.	Dat	е
Inspect Phase Nos.		Inspection Re	quirements	Status	Faults and/or Rema	ırks	Action Taken		Initial
ALL	6.	Access B75R, B90, B120 fuselage turret cavity for cracks, corrosion, and distortion. Inspect for bent or misaligned stringers. Gun area bulkheads for web cracks.							
	7.	Deleted.							
	8.	Deleted.							
ALL C	9.		nd rod ends for dents, curity, and evidence of ds for worn or seized						
ALL C	10.	Flight control bellcrar corrosion and securit mounting security an interference. Pivot be	ty. Brackets for						

PHA	PHASE NO		Area Name and No. MID AND LOWER FUSELAGE – 12				Aircraft Serial No.	Date	е
Inspect Phase Nos.		Inspection Red	quirements	Status	Faults and/or Remar	rks	Action Taken		Initial
2,4 C	11.		ecoupler (ARDD) units , security, evidence of ess and lost motion.						
ALL C	12.	security. Wiring harned connections, chafing	, or deterioration and nce. Rod ends for worn						
2,4 C	13.	Ammo bay interior budistortion, and corrosmisaligned stringers.	sion. Check for bent or						
2,4 C	14.	Ammo bay interior wifor cracks, corrosion, hardware, and securi	loose or missing						
ALL C	15.	Ammo bay interior fu cracks, delamination missing hardware. Fo chafing and security.							

РНА	SE NO	-	a Name and No. DWER FUSELAGE – 12			Aircraft Serial No.	Date	е
Inspect Phase Nos.	Inspection	Requirements	Status	Faults and/or Rema	arks	Action Taken		Initial
2,4 C	structure (transmi	 Ammo bay interior mast strut support structure (transmission deck bottom corners) for cracks, distortion, and security. 						
2,4 C		r flight control servo structure (transmission ard) for cracks, distortion,						
2,4	pressure switches	r fuel pumps and valves, s, manifold, couplings, tings for leakage, cracks, s and security.						
ALL	and drain tubes for corrosion, and se	r fuel lines, hoses, vent, or leakage, chafing, curity. Refueling line les for torque strips and ge.						
2,4		r wiring harnesses for I insulation, and security.						

РНА	PHASE NO MID AND L			a Name and No. OWER FUSELAGE – 12			Aircraft Serial No.	Date	е
Inspect Phase Nos.		Inspection Red	quirements	Status	Faults and/or Rema	rks	Action Taken		Initial
ALL	21.	Ammo bay interior hydraulic lines for leakage, corrosion, mounting security, and chafing.							
2,4	22.		e bolt support brackets and loose or missing olts for damage and						
ALL C	23.	Nitrogen inert composecurity, loose or mis tubes, hoses, and brodamage and security obstructions.	ssing hardware. Inspect eak away values for						
ALL	24.	Ammo bay panel for corrosion, and securi cracks, security, and	ity. Panel latches for						
2,4 C	25.	Main landing gear croscratches, fractures,							

PHA	SEI	NO	Area Name and No. MID AND LOWER FUSELAGE – 12			Aircraft Serial No. Date			е
Inspect Phase Nos.		Inspection Requirements			Faults and/or Rema	rks	Action Taken		Initial
2,4	26.	6. Aft stowage and avionics compartments for cleanliness, cracks, distortion, corrosion, loose or missing rivets or fastners. Seals and copper coat for deterioration and wear.							
2,4	27.	 Fuselage stowage compartments for cracks, distortion, corrosion, and loose or missing rivets and hardware. 							

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PHA	PHASE NO PHASED MAINTENANCE CHECKLIST										
AR	MAN	Area Name and No. MENT MID AND LOWER FUSELAGE – 12		Aircraft Serial No.	Date	Total Hrs. This Area	a				
Inspect Phase Nos.		Inspection Requirements	Status	Faults and/or Remarks	,	Action Taken					
ALL	1.	Index drive, ejector cover, rotor, spacer, and eject guide for cracks, distortion, corrosion, nicks, gouges, and scouring, turn vertical drive shaft and check for binding.									
ALL	2.	Bolt carrier for cracks, nicks, gouges, corrosion, and scouring.									
ALL	3.	Forward track assembly for cracks, nicks, gouges, corrosion, and scouring.									
ALL	4.	Recoil adapter for leakage and physical damage. Clamp halves for corrosion, cracks, and worn clamp pin holes.									
ALL	5.	Blast suppressor for cracks, distortions and proper installation.									
ALL	6.	Barrel for cracks and distortion. Barrel support for cracks.									

РНА	PHASE NO.			a Name a	and No. WER FUSELAGE – 12		Aircraft Serial No.	Date	е
Inspect Phase Nos.		Inspection Re	quirements	Status	Faults and/or Remai	rks	Action Taken		Initial
ALL	7.	Driver motor terminal lug for physical damage and fraying wires. Cable guard for physical damage. Wire support clamps for cracks and wear. Wire harness for chafing. Drive motor for cracks and corrosion.							
ALL	8.	8. Receiver assembly for cracks, distortion, burrs, nicks, gouges, and oil leakage. Turn vertical drive shaft and check for binding and corrosion.							
ALL	9.	Hydraulic drive moto threads for distortion	r tube fittings and , cracks, and leakage.						
ALL	10.	Hydraulic solenoid va bracket for cracks an	alve for leaks. Mounting ad distortion.						
ALL	11.	Hydraulic actuator assembly mounting points for cracks and distortion.							
ALL	Electrical connectors and wires for breaks, fraying and bent or broken connector pins. None allowed.								

РНА	SE NO.			a Name a	and No. WER FUSELAGE – 12		Aircraft Serial No.	Dat	е
Inspect Phase Nos.		Inspection Rec	quirements	Status	Faults and/or Rema	arks	Action Taken		Initial
ALL	broken	 Azimuth drive assembly for cracks, bent or broken brackets. Excessive play or binding bearings and damaged drive gearbox cover. 							
ALL		 Mounting surfaces of fork and dust shield for cracks and distortion. 							
ALL		 Azimuth resolver assembly adapter and spur gear for cracks and distortions. 							
ALL		 Elevation resolver assembly adapter for cracks and distortion. 							
ALL		 Adapter mounting holes in trunnion shaft for cracks and stripped threads. 							
ALL	18. Train ra and dist		unting point for cracks						
ALL	 Turret wiring and connections for breaks, fraying, and broken or bent pins. 								

РНА	SEI	NO		rea Name and No. AND LOWER FUSELAGE – 12			Aircraft Serial No.	Dat	е
Inspect Phase Nos.		Inspection Requirements		Status	Faults and/or Rema	rks	Action Taken		Initial
ALL	20.	Turret mounting area for cracks and corrosion.							
ALL	21.	. Support cradle for bends and distortions.							
ALL	22.	Trunnion bearing for	damage.						
ALL	23.	Support fork for damage, cracks, deep scratches in either leg and corrosion, damaged trunnion and threaded inserts.							
ALL	24.	Gun turret gun cradle support fork shouldered shafts for corrosion, galling and excessive wear.							
ALL	25.	 Fork mounting area on azimuth housing and cradle for cracks, distortion, and stripped threads. 							
ALL	26. Stow spring assembly mounting points for cracks, distortion, and corrosion.								

РНА	PHASE NO.			a Name a	and No. WER FUSELAGE – 12		Aircraft Serial No.	Date	е
Inspect Phase Nos.		Inspection Requirements			Faults and/or Rema	rks	Action Taken		Initial
ALL	27.	 Ammo storage magazine mounts for cracks, damage, and corrosion. 							
ALL	28.	28. Magazine for loose or missing blind fastners, screws, cracks, and dents.							
ALL	29.	 Accelerator/merger assembly for cracks, loose or missing hardware, damage and corrosion. 							
ALL	30.		ly left and right flex mage and corrosion.						
ALL	31.	Carrier drive assemb	ly hydraulic lines and nd leakage.						
ALL	32. Carrier drive assembly connector for cracks, corrosion and bent or broken pins.								
ALL	33. Carrier drive servo manifold for cracks and damage.								

РНА	SEI	NO		ea Name and No. AND LOWER FUSELAGE – 12		Aircraft Serial No.	Dat	е	
Inspect Phase Nos.		Inspection Rec	quirements	Status	Faults and/or Rema	arks	Action Taken		Initial
ALL	34.	 Carrier drive guide support bracket, chute attach bracket, bracket assembly, upper and lower rounds guides for cracks, distortions, corrosion and helicoil inserts for damage. 							
ALL	35.	Ammo conveyor syst corrosion and proper							
ALL	36.	Remove conveyor assembly from aircraft. Check all feeders for cracks, bends, and elongated connecting holes. None allowed. Use ammunition tool kit.							
ALL	37.	7. Area weapon system for evidence of hydraulic fuild leakage. Turret hoses and tubes for leakage, corrosion, and security. Wiring harnesses for loose connections, chafing and deterioration.							
ALL	38. Aft avionics equipment for mounting security, external damage, loose connectors. Wiring harnesses and clamps for chafing and deterioration.								

PHA	SE	NO P	HASED	MAINTENANCE CHECKLIST			
		Area Name and No. TAILBOOM - 13	_	Aircraft Serial No.	Date	Total Hrs. This Area	_
Inspect Phase Nos.		Inspection Requirements	Status	Faults and/or Remarks		Action Taken	Initial
ALL	1.	Exterior skin for holes, dents, cracks, corrosion, loose or working rivets, and loose or missing hardware. Steps for damage, structural integrity, and mounting security.					
ALL	2.	Access panels and fairings for deformation, cracks, corrosion, loose or working rivets, and loose or missing hardware. Fairing, hinges and latches for damage, binding, security, and proper operation. Seals and copper coat for wear or deterioration.					
ALL	3.	Interior components for mounting security and loose or missing hardware.					
ALL	4. Tailboom deck structure for cracks, distortion, corrosion, loose or working rivets, and loose or missing hardware.						

	РНА	SE N	0		a Name a ILBOOM			Aircraft Serial No.	Dat	е
	Inspect Phase Nos.		Inspection Red	quirements	Status	Faults and/or Rema	arks	Action Taken		Initial
1	ALL C	c	Tail rotor drive shaft and couplings for cracks, dents, distortion, corrosion, and evidence of interference.							
	ALL C		Tail rotor drive shaft and coupling bolts for proper torque.							
•	ALL C		Drive shaft dampers for proper friction adjustment.							
	ALL C	c iı	. Tail rotor flight control rods for dents, cracks, corrosion, security, and evidence of interference. Check rod ends for worn or seized bearings.							
	ALL C	c	Tail rotor flight control bellcranks for cracks, corrosion, security, and evidence of interference. Pivot bearing for looseness.							

РНА	PHASE NO			a Name a ILBOON		Aircraft Serial No.		Date	
Inspect Phase Nos.		Inspection Requirements		Status	Faults and/or Rema	rks Action Taken			Initial
2,4		Tailboom interior hydraulic components and lines for leakage, dents, corrosion, chafing and security. Tailboom interior wiring harnesses for chafing, deterioration, and security.							
2,4									
ALL C	12.	Tailboom armor channels for cracks and loose or missing hardware.							
2,4 C		 Tailboom splice (FS 436.5/476.6) for sheared or working rivets, corrosion, and cracked or deformed skin. 							

РНА	PHASE NO PHASED MAINTENANCE CHECKLIST											
E	EMPE	Area Name and No. ENNAGE, TAIL ROTOR, AND TLG – 14		Aircraft Serial No.	Date	Total Hrs. This Area	l					
Inspect Phase Nos.		Inspection Requirements	Status	Faults and/or Remarks		Action Taken	Initial					
ALL	1.	CAUTION: DO NOT USE TAIL ROTOR PUSHRODS AS HANDHOLDS. Exterior skin for holes, dents, cracks, corrosion, loose or working rivets, and loose or missing hardware. Steps for damage, structural integrity, and mounting security.										
ALL	2.	Access panels and fairings for deformation, cracks, corrosion, loose or working rivets, and loose or missing hardware. Lanyards for security and condition. Check for worn or deteriorated seals and copper coat.										
ALL	3.	Structural flanges and fairings on vertical stabilizer for worn, non-adhering, or missing chafe tape.										
ALL	4.	Interior structure for cracks, corrosion, loose or working rivets, and loose or missing hardware.										

РНА	SE	NO		ea Name and No. TAIL ROTOR, AND TLG – 14			Aircraft Serial No.	Date	е
Inspect Phase Nos.		Inspection Red	quirements	Status	Faults and/or Remar	ks	Action Taken		Initial
2,4 C	5.	 Vertical stabilizer elastomeric mounts for cracks and distortion. Mounting bolts and barrel nuts for damage and security. Drain cavity for debris and obstructions. 							
ALL C	5a. Vertical stabilizer mounting for security.								
2,4 C	6. Stabilizer navigation and formation lights for corrosion, loose or missing hardware, and security. Lenses for cracks, security, and discoloration. Wiring harnesses for loose connections and chafing. (Remove R578).								
2,4	7.	7. Stabilizer radar warning antennas, GPS antenna, and FM-AM whip antenna for damage, mounting security, and condition of wiring. (Remove T545).							
2,4 C	8. Stabilizer spar box for cracks, corrosion, distortion, and loose or working rivets. (Access L530, L550).								
2,4 C	9. Hydraulic lines and couplings for leaks, cracks, chafing, and clamping security.								

"FOD REMINDER" Check work area for tools and parts after completion of maintenance and inspections.

2-108 C1

PHA	SE I	NO		a Name AIL RO	and No. TOR, AND TLG – 14		Aircraft Serial No.	Dat	е
Inspect Phase Nos.		Inspection Red	quirements	Status	Faults and/or Rema	rks	Action Taken		Initial
2,4 C	10.	Stabilizer trailing edge fairing for damage and corrosion.							
ALL C	Stabilizer structure and skin for cracks, dents, distortion, and loose or missing rivets and hardware.								
ALL C	12.	Stabilator tip fairings for cracks and loose or missing screws.							
ALL C	13.	and corrosion. Wiring for chafing, deterioration, and connection security. Stabilator and aircraft fittings for cracks, loose or missing hardware, and worn or seized bearings.							
ALL C	14.								

PHA	SE NO	Area Name and No. EMPENNAGE, TAIL ROTOR, AND TLG – 14				Aircraft Serial No.	Date	е
Inspect Phase Nos.	Inspection Re	equirements	Status	Faults and/or Rema	rks	Action Taken		Initial
ALL C	pivot bosses for crac corrosion, and clean	5. Stabilator pivot bolts, pivot bearings, and pivot bosses for cracks, distortion, corrosion, and cleanliness. Pivot bearings for looseness. (Remove stabilator).						
ALL C	output retainers for o security. Check for g evidence of overhea	output retainers for cracks, distortion, and security. Check for grease leakage and evidence of overheating (discoloration). Gearbox mount fittings for cracks and distortion. Intermediate gearbox mounting bolts for proper torque.						
2,4 C								
ALL C	and defuser for crac Impeller for cracks a							
ALL C	 Intermediate gearbox thermistors for insulation damage, grease leakage, and security. Wire harness splices for security. 							

РНА	SEI	NO			a Name and No. AIL ROTOR, AND TLG – 14		Aircraft Serial No.	Dat	е
Inspect Phase Nos.		Inspection Re	quirements	Status	Faults and/or Rema	rks	Action Taken		Initial
2,4	20.	Intermediate gearbox wiring harness for loose connections, chafing, or deterioration.							
ALL C	21.	21. Tail rotor drive shaft and couplings for cracks, dents, distortion, and corrosion.							
ALL C	22.	22. Tail rotor drive shaft and coupling bolts, verify torque.							
ALL C	23.	 Tail rotor gearbox brace for loose, worn or seized bearings, cracked or broken fittings, cracked strut, and loose or missing hardware. 							
ALL C	24.	Tail rotor gearbox ho output retainers for consecurity. Check for good evidence of overhead	racks, distortion, rease leakage and						
ALL	25.	Tail rotor gearbox mo	ount fittings for cracks						

РНА	SE I	NO		a Name a AIL RO	and No. TOR, AND TLG – 14		Aircraft Serial No.	Dat	е
Inspect Phase Nos.		Inspection Rec	quirements	Status	Faults and/or Rema	ırks	Action Taken		Initial
ALL C	26.	Tail rotor gearbox mounting studs, verify torque.							
2,4	27.	Tail rotor gearbox thermistors for insulation damage, grease leakage, and security. Wire harness splices for security, loose connections, chafing and deterioration.							
ALL C	28.	Tail rotor flight control rods for dents, cracks. corrosion, security, and evidence of interference. Check rod ends for worn or seized bearings.							
ALL C	29.	Tail rotor pitch chang cracks, corrosion, se interference, worn or tolerance wear limits.	curity, evidence of seized bearings, and						
2,4 C	30.	. Tail rotor flight control bellcranks for cracks, corrosion, security, and evidence of interference. Pivot bearings for looseness.							

PHA	PHASE NO EMPENNAGE, 1		ea Name and No. ΓAIL ROTOR, AND TLG – 14			Aircraft Serial No.	Dat	е	
Inspect Phase Nos.		Inspection Re	quirements	Status	Faults and/or Rema	rks	Action Taken		Initial
ALL C	31.	31. Tail rotor flight control servo cylinder for leakage, cracks, and corrosion. Rod end and base clevis for bearing damage and attachment security. Rod end for looseness. Rubber boot for cuts, tears, and deterioration. Inspect servo cylinder control linkage hardware for damage and security.							
ALL C	32.	Tail rotor swashplate corrosion. Looseness and rotating swashpl swashplate.	s between stationary						
ALL C	33.		ity. Attaching studs for ee play measurement						
ALL C	34.	Tail rotor fork yoke te cracks, separation, a Bearing center studs looseness.	and deterioration.						

РНА	SE	NO		FAIL ROTOR, AND TLG – 14			Aircraft Serial No.	Date	е
Inspect Phase Nos.		Inspection Red	quirements	Status	Faults and/or Rema	rks	Action Taken		Initial
ALL	35.	5. Tail rotor fork for proper indexing, three studs from the tail rotor gearbox output shaft for wear and three nuts found on those studs for torque. Torque to 60 foot-pounds.							
ALL C	36.	Tail rotor root blade bolt heads and nuts for cracks and looseness. (PLI Method)							
ALL C	37.	Tail rotor blade spars and root finger doublers for delamination.							
ALL C	38.	Tail rotor blade leading dents, distortion, and tips for loose or miss	l erosion. Leading edge						
ALL C	39.	Tail rotor blade bushi for looseness. Dust b misalignment, and se							

РНА	SE	NO		a Name AIL RO	and No. TOR, AND TLG – 14		Aircraft Serial No.	Date	е
Inspect Phase Nos.		Inspection Red	quirements	Status	Faults and/or Rema	rks	Action Taken		Initial
ALL	40.	and security. Pivot pin, bolts, and bushings for looseness.							
ALL	41. Tail landing gear wheel lock actuator for leakage, cracks, and security. Check locking handle and linkage for free movement, full travel, and lateral looseness.								
ALL C	42.	Tail landing gear wiring connections, chafing							
2,4 C	43.	Repack Tail Landing	Gear wheel bearings.						
ALL C	44.	Tail landing gear sho cracks, distortion, an for worn or seized be	d corrosion. Rod ends						
2,4 C	cracks, corrosion, worn or seized bearings								

РНА	SE	NO P	HASE	D MAINTENANCE CHECKLIST			
POST	со	Area Name and No. NDITIONS AND POWER ON CHECKS – 15		Aircraft Serial No.	Date	Total Hrs. This Area	l
Inspect Phase Nos.		Inspection Requirements	Status	Faults and/or Remarks	,	Action Taken	
ALL	1.	Install mast mounted assembly. (if required).					
	l ''	motali madi modifica addenisty. (ii requirea).					
С							
ALL	2.	Install main rotor blades.					
, , , ,		motali malii rotor biadoo.					
	_						
ALL	3.	Install conveyor assembly.					
ALL	4.	Install gun turret assembly.					
ALL	5.	Install ammo magazine.					
ALL	J.	mstall animo magazino.					
ALL	6.	Lubricate ammunition handling subsystem.					ļ
ALL	7.	Adjust ammunition handling subsystem.					
A. I.	٥	Deposed singreft					
ALL	8.	Repanel aircraft.					
ALL	9.	Install store jettison cartridges.					

РНА	SE N	NO		a Name a AND PO	and No. DWER ON CHECKS – 15		Aircraft Serial No.	Date	е
Inspect Phase Nos.		Inspection Re	quirements	Status	Faults and/or Remar	ks	Action Taken		Initial
ALL C	10.	Start APU.							
ALL C	11.	11. Perform ARDD 250 hour inspection.							
ALL C	11a. Perform 250 hour scheduled inspection: BUCS servocylinders.								
ALL C	12. Flight controls actuated through full stop-to-stop ranges. Main and tail rotor pitch change functions for full travel, control response, binding, looseness and noisy operation.								
ALL C	13.	Perform battery char	ger operational check.						
ALL C	14.	Operate fuel boost polight.	ump for fuel pressure						
ALL C	15.	Perform fuel system	leak-check.						
ALL C	Stabilator actuated through full travel range for smooth operation with no lost motion or binding.								

"FOD REMINDER" Check work area for tools and parts after completion of maintenance and inspections. 2-118 C1

РНА	PHASE NO POST CONDITIONS			a Name a AND PO	and No. DWER ON CHECKS – 15		Aircraft Serial No.	Date	е
Inspect Phase Nos.		Inspection Red	quirements	Status	Faults and/or Rema	rks	Action Taken		Initial
ALL C	17.	 Forward and aft avionics bay cooling fans for smooth operation. 							
ALL C	18.	Perform windshield a operational check.	anti-ice system						
ALL C	19.	Operate rotor brake to operational modes.	for brake, lock, and off						
ALL C	20.	Perform pitot heat ch	eck.						
ALL C	21.	Perform tail rotor and alarm 250 hour inspe	d intermediate gearbox ection						
	22.	Deleted.							
ALL C	23.	Start both engines.							

РНА	PHASE NO POST CONDITIONS			a Name a AND PC	and No. DWER ON CHECKS – 15		Aircraft Serial No.	Date	е
Inspect Phase Nos.		Inspection Red	quirements	Status	Faults and/or Remai	rks	Action Taken		Initial
ALL C	24.	24. Perform HIT checks on both engines.							
ALL C	25.	 Perform drives MOC for main transmisson and nose gearbox transducers, temperature probes, and oil pressure transducers. 							
ALL C	26.	 Operate engine anti-ice system. Verify rise in engine TGT. 							
ALL C	27.		uate air flow and cold outputs in both pilot						
ALL C	28.	Perform engine chop 1-1520-251-MTF.	operational check. TM						
ALL C	29.		e been completed or ms initiated as e been carried forward 13 or DA Form 2408-14						

PHA	SE NO			a Name AND PO	and No. OWER ON CHECKS – 15		Aircraft Serial No.	Date	
Inspect Phase Nos.		Inspection Re	quirements	Status	Faults and/or Rema	rks	Action Taken		Initial
ALL C	oper: compaction funct be per the eaction oper: be per Copii 4676	operational checks (MOC). After the completion of any required corrective actions to any of the components of a functional system of the aircraft. MOCs shall be performed on the system to determine the effectiveness of the maintenance actions performed and to verify the proper operation of the system. These MOC's shall be performed in accordance with the IETM. Copies of supplemental sheets (DA Form 4676-R) may be used to record and sign off the maintenance operational checks performed.							
ALL C	opers Perfo comp durin	ational checks (I orm MOC's on a conents that having the inspection	tion maintenance MOC), as required. ircraft systems or we been disturbed h. MOC's shall be ance with the IETM.						
ALL C	32. Perfo	orm 10 hour/14 o	day inspection.						

PHA	DILLAGE NO.		a Name and No. AND POWER ON CHECKS – 15			Aircraft Serial No.	Date	е	
Inspect Phase Nos.		Inspection Requirements		Status	Faults and/or Rema	arks Action Taken			Initial
ALL C		3. Perform post-inspection MTF in accordance with TM 1-1520-251-MTF and TM 1-1520-328-23.							
	34. Deleted.								

РНА	SE NO P	PHASED MAINTENANCE CHECKLIST							
	Area Name and No. FINAL INSPECTION REQUIREMENTS	Aircraft Serial No.		Date	Total Hrs. This Area	1			
Inspect Phase Nos.	Inspection Requirements	Status	Faults and/or Remarks		Action Taken	Initial			
ALL C	Final records check for completion of AH-64D helicopter phased maintenance inspection.								
ALL C	Complete final records check in automated phase manual.								
	3. Deleted.								
	4. Deleted.								

By Order of the Secretary of the Army:

OFFICIAL:

ERIC K. SHINSEKI General, United States Army Chief of Staff

JOEL B. HUDSON Administrative Assistant to the Secretary of the Army 0210101

Joel B. Hula

DISTRIBUTION:

To be distributed in accordance with Initial Distribution Number (IDN 314046) requirements for TM 1-1520-251-PM.

These are the instructions for sending an electronic 2028

The following format must be used if submitting an electronic 2028. The subject line must be exactly the same and all fields must be included; however only the following fields are mandatory: 1, 3, 4, 5, 6, 7, 8, 9, 10, 13, 15, 16, 17, and 27.

From: "Whomever" < whomever@wherever.army.mil>

To: 2028@redstone.army.mil

Subject: DA Form 2028

1. *From:* Joe Smith

2. Unit: home

3. *Address:* 4300 Park4. *City:* Hometown

5. **St:** MO6. **Zip:** 77777

7. Date Sent: 19-OCT-93
 8. Pub no: 55-2840-229-23

9. **Pub Title:** TM

10. Publication Date: 04-JUL-85

11. Change Number: 7
12. Submitter Rank: MSG
13. Submitter FName: Joe
14. Submitter MName: T
15. Submitter LName: Smith

16. **Submitter Phone:** 123–123–1234

17. **Problem: 1**18. Page: 2
19. Paragraph: 3

20. Line: 4 21. NSN: 5 22. Reference: 6 23. Figure: 7 24. Table: 8 25. Item: 9

27. **Text**:

26. Total: 123

This is the text for the problem below line 27.

RECOMMENDED CHANGES TO PUBLICATIONS AND BLANK FORMS

For use of this form, see AR 25-30; the proponent agency is ODISC4.

Use Part II (reverse) for Repair Parts and Special Tool Lists (RPSTL) and Supply Catalogs/ Supply Manuals (SC/SM)

DATE

8/30/02

TO: (Forward to proponent of publication or form)(Include ZIP Code)

Commander, U.S. Army Aviation and Missile Command

ATTN: AMSAM-MMC-MA-NP Redstone Arsenal, AL 35898 FROM: (Activity and location)(Include ZIP Code)

MSG, Jane Q. Doe 1234 Any Street

Nowhere Town, AL 34565

PART 1 - ALL PUBLICATIONS (EXCEPT RPSTL AND SC/SM) AND BLANK FORMS

DUDUGA	TION:/FO			OBLICATI	.5140 (LA		TITLE Organizational Direct Company And Consul				
	9–100	7M NUMBER 5–433–2				16 Sep 2002	TITLE Organizational, Direct Support, And General Support Maintenance Manual for Machine Gun, .50 Caliber M3P and M3P Machine Gun Electrical Test Set Used On Avenger Air Defense Weapon System				
ITEM NO.	PAGE NO.	PARA- GRAPH	LINE NO. *	FIGURE NO.	TABLE NO.	RECO	DMMENDED CHANGES AND REASON				
1	WP0005 PG 3		2			Test or Corrective Ac	t or Corrective Action column should identify a different WP number.				
				•	1						
			1	P							

* Reference to line numbers within the paragraph or subparagraph.

MSG, Jane Q. Doe, SFC

TYPED NAME, GRADE OR TITLE

TELEPHONE EXCHANGE/ AUTOVON, PLUS EXTEN-SION

788-1234

SIGNATURE

Comma	ander, U. AMSAM-	S. Army -MMC-N	ddressee listed in publication Aviation and Missile Communication MA-NP AL 35898	FROM: (Activity and location) (Include ZIP Code) MSG, Jane Q. Doe 1234 Any Street Nowhere Town, AL 34565							8/30/02	
PUBLICATION NUMBER						LISTS AN	ID SUPI	PLY CATALOGS/SUPPLY MANUALS TITLE			S	
PAGE NO.	COLM LINE NATIONAL STOCK NO. NO. NUMBER			RENCE O.	FIGURE NO.	ITEM NO.	TOTAL OF MA ITEI SUPPO	AJOR MS	RECOMMEN		DED ACTION	
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			blank forms	"ional b	la). he	eis nay be	used if I	more spa	nce is nee	eded.)		
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MSG Jane Q Doe SEC					788	3-123						

RE		BLA	NK FORM	PUBLICATION IS onent agency is ODISC			Use Part II (reverse) for Repair Parts and Special Tool Lists (RPSTL) and Supply Catalogs/ Supply Manuals (SC/SM)				
Comman ATTN: A	nder, U.S. A	Army Aviation MC-MA-NP	ublication c	or form)(Includille ile Command	de ZIP Cod	de)	P) FROM: (Activity and location)(Include ZIP Code)				
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PUBLICATION/FORM NUMBER								TITLE			
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ATTN:	orward dir lander, U. : AMSAN one Arser	1-MMC-N		on) I nand	FROM: (Activity and location) (Include ZIP Code) DATE						
		PART	II - REPAIR PARTS AND	SPECIAL	TOOL	LISTS AN	ID SUP	PLY CATAL	OGS/SUPPLY	/ MANUALS	
PUBLIC	CATION N	\UMBEF	₹]	DATE			TITLE			
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	PAF	RT III – F	REMARKS (Any general re	emarks or	recomi	mendations	, or sug	gestions for	<i>improvement</i> o	of publication	ns and
			blank forms. Add	litional bla	nk she	ets may be	used if I	more space	is needed.)		
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The Metric System and Equivalents

Linear Measure

Liquid Measure

- 1 centimeter = 10 millimeters = .39 inch
- 1 decimeter = 10 centimeters = 3.94 inches
- 1 meter = 10 decimeters = 39.37 inches
- 1 dekameter = 10 meters = 32.8 feet
- 1 hectometer = 10 dekameters = 328.08 feet
- 1 kilometer = 10 hectometers = 3,280.8 feet

Weights

- 1 centigram = 10 milligrams = .15 grain
- 1 decigram = 10 centigrams = 1.54 grains
- 1 gram = 10 decigrams = .035 ounce
- 1 dekagram = 10 grams = .35 ounce
- 1 hectogram = 10 dekagrams = 3.52 ounces
- 1 kilogram = 10 hectograms = 2.2 pounds
- 1 quintal = 100 kilograms = 220.46 pounds
- 1 metric ton = 10 quintals = 1.1 short tons

- 1 centiliter = 10 milliliters = .34 fl. ounces
- 1 deciliter = 10 millillers = .34 ft. ounces
- 1 liter = 10 deciliters = 33.81 fl. ounces
- 1 dekaliter = 10 liters = 2.64 gallons
- 1 hectoliter = 10 dekaliters = 26.42 gallons
- 1 kiloliter = 10 hectoliters = 264.18 gallons

Square Measure

- 1 sq. centimeter = 100 sq. millimeters = .155 sq. inch
- 1 sq. decimeter = 100 sq. centimeters = 15.5 sq. inches
- 1 sq. meter (centare) = 100 sq. decimeters = 10.76 sq. feet
- 1 sq. dekameter (are) = 100 sq. meters = 1,076.4 sq. feet
- 1 sq. hectometer (hectare) = 100 sq. dekameters = 2.47 acres
- 1 sq. kilometer = 100 sq. hectometers = .386 sq. mile

Cubic Measure

- 1 cu. centimeter = 1000 cu. millimeters = .06 cu. inch
- 1 cu. decimeter = 1000 cu. centimeters = 61.02 cu. inches
- 1 cu. meter = 1000 cu. decimeters = 35.31 cu. feet

Approximate Conversion Factors

To change	To	Multiply by	To change	To	Multiply by
inches	centimeters	2.540	ounce-inches	newton-meters	.007062
feet	meters	.305	centimeters	inches	.394
yards	meters	.914	meters	feet	3.280
miles	kilometers	1.609	meters	yards	1.094
square inches	square centimeters	6.452	kilometers	miles	.621
square feet	square meters	.093	square centimeters	square inches	.155
square yards	square meters	.836	square meters	square feet	10.764
square miles	square kilometers	2.590	square meters	square yards	1.196
acres	square hectometers	.405	square kilometers	square miles	.386
cubic feet	cubic meters	.028	square hectometers	acres	2.471
cubic yards	cubic meters	.765	cubic meters	cubic feet	35.315
fluid ounces	milliliters	29.573	cubic meters	cubic yards	1.308
pints	liters	.473	milliliters	fluid ounces	.034
quarts	liters	.946	liters	pints	2.113
gallons	liters	3.785	liters	quarts	1.057
ounces	grams	28.349	liters	gallons	.264
pounds	kilograms	.454	grams	ounces	.035
short tons	metric tons	.907	kilograms	pounds	2.205
pound-feet	newton-meters	1.356	metric tons	short tons	1.102
pound-inches	newton-meters	.11296			

Temperature (Exact)

PIN: 073278